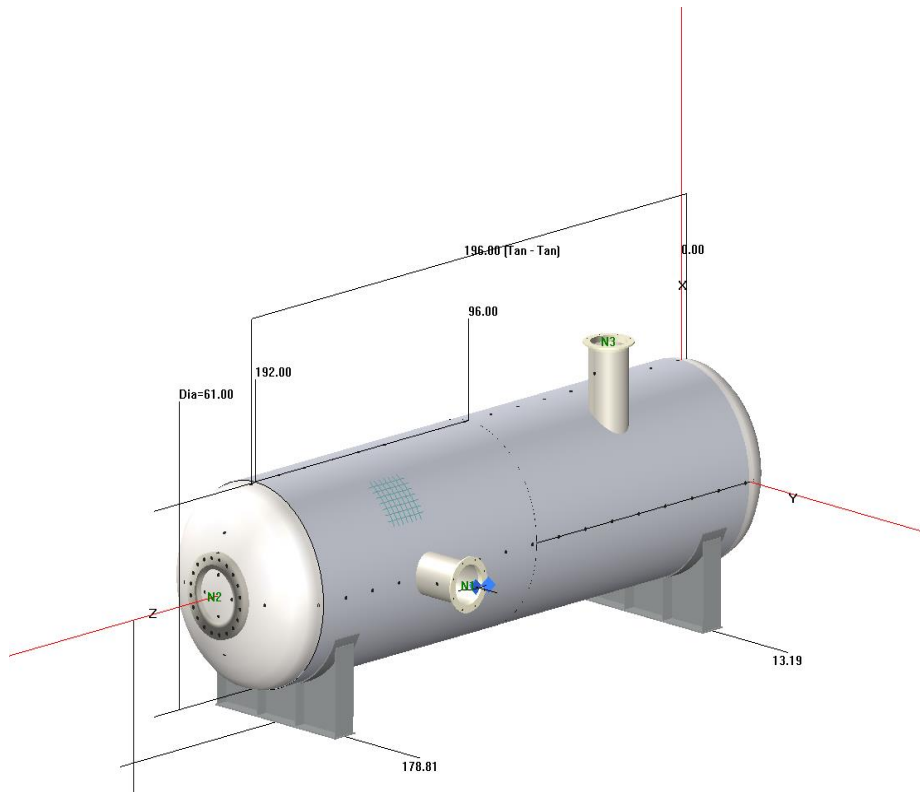


Codeware, Inc.

Sarasota, FL, USA

www.codeware.com



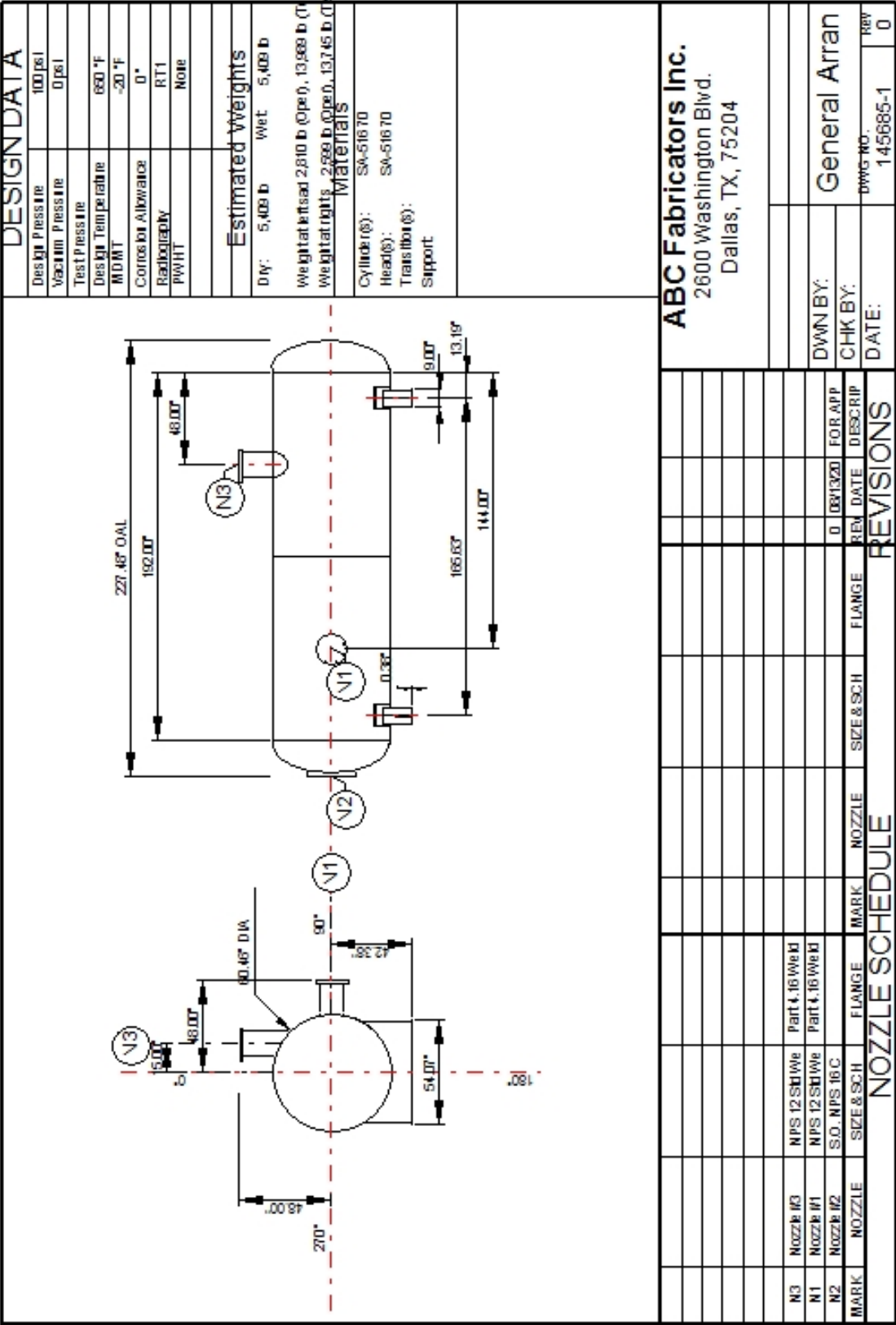
INSPECT Pressure Vessel Design Calculations

Item: Split Stream Dearator
Vessel No: V-1234
Customer: Magaladon Oil Venture
Contract: C-45490-R56
Designer: John Doe
Date: May 14, 2009

You can edit this page by selecting **Cover Page settings...** in the **report** menu.

Table of Contents

<u>General Arrangement Drawing</u>	1/98
<u>Deficiencies Summary</u>	2/98
<u>Nozzle Schedule</u>	3/98
<u>Nozzle Summary</u>	4/98
<u>Pressure Summary</u>	5/98
<u>Revision History</u>	6/98
<u>Settings Summary</u>	7/98
<u>Radiography Summary</u>	9/98
<u>Thickness Summary</u>	10/98
<u>Weight Summary</u>	11/98
<u>Long Seam Summary</u>	12/98
<u>Test Report</u>	13/98
<u>Inspection on 12/12/2013</u>	14/98
<u>Cylinder #1</u>	23/98
<u>Cylinder #2</u>	25/98
<u>Ellipsoidal Head #1</u>	27/98
<u>Straight Flange on Ellipsoidal Head #1</u>	29/98
<u>Ellipsoidal Head #1 (Dished Region)</u>	31/98
<u>Straight Flange on Ellipsoidal Head #2</u>	33/98
<u>Ellipsoidal Head #2</u>	35/98
<u>Ellipsoidal Head #2 (Dished Region)</u>	37/98
<u>Flange #2</u>	39/98
<u>Flange #3</u>	47/98
<u>Nozzle #1 (N1)</u>	55/98
<u>Nozzle #2 (N2)</u>	74/98
<u>Nozzle #3 (N3)</u>	79/98
<u>Saddle #1</u>	89/98



Deficiencies Summary

Deficiencies for [Nozzle #1 \(N1\)](#)

Leg₄₁ = 0.25 in is less than minimum required leg size of 0.3571 in.

Leg₄₂ = 0.1875 in is less than minimum required leg size of 0.2143 in.

Deficiencies for [Nozzle #2 \(N2\)](#)

Shell opening diameter (16.3372 in) must be larger than 16.46 in [stud pad inner diameter (16 in) plus 2 times the lower fillet weld leg (0.23 in)].

Deficiencies for [Nozzle #3 \(N3\)](#)

Leg₄₁ = 0.25 in is less than minimum required leg size of 0.3571 in.

Leg₄₂ = 0.1875 in is less than minimum required leg size of 0.2143 in.

Warnings Summary

Warnings for [Cylinder #1](#)

INSPECT only supports butt joint construction in shell components. Please check the circumferential joint between Cylinder #1 and Cylinder #2. A common source of this error is specifying one component as an OD design and an adjacent component as an ID design. (warning)

Warnings for [Cylinder #2](#)

INSPECT only supports butt joint construction in shell components. Please check the circumferential joint between Cylinder #2 and Ellipsoidal Head #2. A common source of this error is specifying one component as an OD design and an adjacent component as an ID design. (warning)

Warnings for [Ellipsoidal Head #1](#)

INSPECT only supports butt joint construction in shell components. Please check the circumferential joint between Ellipsoidal Head #1 and Cylinder #1. A common source of this error is specifying one component as an OD design and an adjacent component as an ID design. (warning)

Warnings for [Nozzle #1 \(N1\)](#)

Leg₄₁ = 0.25 in is less than minimum required leg size of 0.3571 in. (new condition) (warning)

Leg₄₂ = 0.1875 in is less than minimum required leg size of 0.2143 in. (new condition) (warning)

The design of this nozzle does not include mill undertolerance because a minimum (as built) thickness was entered. (warning)

Warnings for [Nozzle #3 \(N3\)](#)

Leg₄₁ = 0.25 in is less than minimum required leg size of 0.3571 in. (new condition) (warning)

Leg₄₂ = 0.1875 in is less than minimum required leg size of 0.2143 in. (new condition) (warning)

The design of this nozzle does not include mill undertolerance because a minimum (as built) thickness was entered. (warning)

Nozzle Schedule

Nozzle mark	Service	Size	Materials		Impact Tested	Normalized	Fine Grain	Flange	Blind
N1	Nozzle #1	NPS 12 Std Weight	Nozzle	SA-106 B Smls pipe	No	No	No	Part 4.16 Weld Neck Integral SA-516 70	No
			Pad	SA-516 70	No	No	No		
N2	Nozzle #2	Studding Outlet NPS 16 Class 300	Nozzle	SA-105	No	No	No	N/A	No
N3	Nozzle #3	NPS 12 Std Weight	Nozzle	SA-106 B Smls pipe	No	No	No	Part 4.16 Weld Neck Integral SA-516 70	No
			Pad	SA-516 70	No	No	No		

Nozzle Summary

Nozzle mark	OD (in)	t (in)	Req t (in) ⁿ	Shell			Reinforcement Pad		Corr (in)	Gov. Stress Ratio (%)
				Nom t (in)	Design t (in)	User t (in)	Width (in)	t _{pad} (in)		
N1	12.75	0.375	0.3281	0.371	N/A	0.371	3	0.25	0	100.0
N2	25.5	4.75	0.2556	0.23*	0.207	0.23	N/A	N/A	0	119.6
N3	12.75	0.375	0.3281	0.37	N/A	0.37	3	0.25	0	100.0

t_n : Nozzle thickness
 Req t_n : Nozzle thickness required per 4.5.4.1
 Nom t: Vessel wall thickness
 Design t: Required vessel wall thickness due to pressure + corrosion allowance per 4.5.17
 User t: Local vessel wall thickness (near opening)
 Gov. Stress Ratio: Minimum stress ratio, S_{allow} / P_L , S / τ , S / τ_n
 Corr: Corrosion allowance on nozzle wall
 *: Head minimum thickness after forming

Pressure Summary

Pressure Summary for Chamber bounded by Ellipsoidal Head #2 and Ellipsoidal Head #1

Identifier	P Design (psi)	T Design (°F)	MAWP (psi)	MDMT (°F)	MDMT Exemption		Impact Tested
Ellipsoidal Head #1	100	650	144.02	N/A	N/A		No
Ellipsoidal Head #1 (Dished Region)	100	650	160.69	N/A	N/A		No
Straight Flange on Ellipsoidal Head #1	100	650	233.25	N/A	N/A		No
Cylinder #1	100	650	230.08	N/A	N/A		No
Cylinder #2	100	650	230.12	N/A	N/A		No
Straight Flange on Ellipsoidal Head #2	100	650	233.25	N/A	N/A		No
Ellipsoidal Head #2	100	650	144.02	N/A	N/A		No
Ellipsoidal Head #2 (Dished Region)	100	650	160.69	N/A	N/A		No
Saddle #1	100	650	144.02	N/A	N/A		N/A
Nozzle #1 (N1)	100	650	175.89	N/A	Nozzle	N/A	No
					Pad	N/A	No
Flange #2	100	650	166.57	N/A	N/A		No
Nozzle #2 (N2)	100	650	160.01	N/A	N/A		No
Nozzle #3 (N3)	100	650	176.02	N/A	Nozzle	N/A	No
					Pad	N/A	No
Flange #3	100	650	166.57	N/A	N/A		No

Chamber MAWP hot & corroded is 144.02 psi @ 650 °F

This pressure chamber is not designed for external pressure.

Design notes are available on the [Settings Summary](#) page.

Revision History

No.	Date	Operator	Notes
0	8/13/2013	Matt	New vessel created ASME Section VIII Division 1 [COMPRESS 2013 Build 7330]
1	12/13/2013	Matt	Converted from ASME Section VIII Division 1, 2010 Edition, A11 Addenda to ASME Section VIII Division 1, 2013 Edition. During the conversion, changes may have been made to your vessel (some may be listed above). Please check your vessel carefully.

Settings Summary

INSPECT 2014 Build 7400

Vessel calculations are based on the projected corroded state of the vessel at the date of the inspection on 12/12/2013

Units: U.S. Customary

Datum Line Location: 0.00" from right seam

Design

ASME Section VIII Division 1, 2013 Edition

Design or Rating:	Get Pressure Rating and Calculate Required Thickness
Minimum thickness:	0.0625" per UG-16(b)
Design for cold shut down only:	No
Design for lethal service (full radiography required):	No
Rate nozzles for:	Find nozzle MAWP
Corrosion weight loss:	100% of theoretical loss
UG-23 Stress Increase:	1.20
Skirt/legs stress increase:	1.0
Minimum nozzle projection:	0.001"
Juncture calculations for $\alpha > 30$ only:	Yes
Preheat P-No 1 Materials $> 1.25"$ and $\leq 1.50"$ thick:	No
UG-37(a) shell tr calculation considers longitudinal stress:	No
Pipe under-tolerance is not applied to cylindrical shell thicknesses.	
Pipe under-tolerance is not applied to nozzle wall thicknesses.	
Pipe under-tolerance is not applied to pipe cap thicknesses.	
Butt welds are tapered per Figure UCS-66.3(a).	

Hydro/Pneumatic Test

Maximum stress during test: 90% of yield

Required Marking - UG-116

UG-116(e) Radiography: RT1
UG-116(f) Postweld heat treatment: None

Code Cases/Interpretations

Use Code Case 2547:	No
Use Code Case 2695:	Yes
Apply interpretation VIII-1-83-66:	Yes
Apply interpretation VIII-1-86-175:	Yes
Apply interpretation VIII-1-01-37:	Yes
No UCS-66.1 MDMT reduction:	No
No UCS-68(c) MDMT reduction:	No
Disallow UG-20(f) exemptions:	No

UG-22 Loadings

UG-22(a) Internal or External Design Pressure :	Yes
UG-22(b) Weight of the vessel and normal contents under operating or test conditions:	Yes
UG-22(c) Superimposed static reactions from weight of attached equipment (external loads):	No
UG-22(d)(2) Vessel supports such as lugs, rings, skirts, saddles and legs:	Yes
UG-22(f) Wind reactions:	No
UG-22(f) Seismic reactions:	No
UG-22(j) Test pressure and coincident static head acting during the test:	No
Note: UG-22(b),(c) and (f) loads only considered when supports are present.	

Radiography Summary

Radiography for Chamber bounded by Ellipsoidal Head #2 and Ellipsoidal Head #1

Component	Longitudinal Seam		Left Circumferential Seam		Right Circumferential Seam		Mark
	Category (Fig UW-3)	Radiography / Joint Type	Category (Fig UW-3)	Radiography / Joint Type	Category (Fig UW-3)	Radiography / Joint Type	
Ellipsoidal Head #1	N/A	Seamless No RT	N/A	N/A	B	Full UW-11(a) / Type 1	RT1
Cylinder #1	A	Full UW-11(a) / Type 1	B	Full UW-11(a) / Type 1	B	Full UW-11(a) / Type 1	RT1
Cylinder #2	A	Full UW-11(a) / Type 1	B	Full UW-11(a) / Type 1	B	Full UW-11(a) / Type 1	RT1
Ellipsoidal Head #2	N/A	Seamless No RT	B	Full UW-11(a) / Type 1	N/A	N/A	RT1
Nozzle	Longitudinal Seam		Nozzle to Vessel Circumferential Seam		Nozzle free end Circumferential Seam		
Nozzle #2 (N2)	N/A	Seamless No RT	D	N/A / Type 7	N/A	N/A	N/A
Nozzle #1 (N1)	N/A	Seamless No RT	D	N/A / Type 7	C	Full UW-11(a) / Type 1	RT1
Nozzle #3 (N3)	N/A	Seamless No RT	D	N/A / Type 7	C	Full UW-11(a) / Type 1	RT1
Nozzle Flange	Longitudinal Seam		Flange Face		Nozzle to Flange Circumferential Seam		
Flange #2	N/A	Seamless No RT	N/A	N/A / Gasketed	C	Full UW-11(a) / Type 1	RT1
Flange #3	N/A	Seamless No RT	N/A	N/A / Gasketed	C	Full UW-11(a) / Type 1	RT1
Chamber bounded by Ellipsoidal Head #2 and Ellipsoidal Head #1 - UG-116(e) Radiography: RT1							

Thickness Summary

Component Identifier	Material	Diameter (in)	Length (in)	Nominal t (in)	Design t (in)	Total Corrosion (in)	Joint E	Load
Ellipsoidal Head #1	SA-516 70	60 ID	15.23	0.23*	0.1597	0	1.00	Internal
Straight Flange on Ellipsoidal Head #1	SA-516 70	60 ID	2	0.375	0.1601	0	1.00	Internal
Ellipsoidal Head #1 (Dished Region)	SA-516 70	107.5463 ID	54.0031	0.23*	0.1431	0	1.00	Internal
Cylinder #1	SA-516 70	60.258 ID	96	0.371	0.1607	0	1.00	Internal
Cylinder #2	SA-516 70	60.085 ID	96	0.37	0.1602	0	1.00	Internal
Straight Flange on Ellipsoidal Head #2	SA-516 70	60 ID	2	0.375	0.1601	0	1.00	Internal
Ellipsoidal Head #2	SA-516 70	60 ID	15.23	0.23*	0.1597	0	1.00	Internal
Ellipsoidal Head #2 (Dished Region)	SA-516 70	107.5463 ID	54.0031	0.23*	0.1431	0	1.00	Internal

Nominal t: Vessel wall nominal thickness

Design t: Required vessel thickness due to governing loading + corrosion

Joint E: Longitudinal seam joint efficiency

* Head minimum thickness after forming

Load

internal: Circumferential stress due to internal pressure governs

external: External pressure governs

Wind: Combined longitudinal stress of pressure + weight + wind governs

Seismic: Combined longitudinal stress of pressure + weight + seismic governs

Weight Summary

Component	Weight (lb) Contributed by Vessel Elements										Surface Area ft²
	Metal New*	Metal Corroded*	Insulation	Insulation Supports	Lining	Piping + Liquid	Operating Liquid		Test Liquid		
							New	Corroded	New	Corroded	
Ellipsoidal Head #1	279.5	279.5	0	0	0	0	0	0	1,232.1	1,232.1	30
Cylinder #1	1,906.4	1,906.4	0	0	0	0	0	0	9,953.9	9,953.9	127
Cylinder #2	1,895.8	1,895.8	0	0	0	0	0	0	9,913.7	9,913.7	127
Ellipsoidal Head #2	312.8	312.8	0	0	0	0	0	0	1,224.7	1,224.7	33
Saddle #1	578	578	0	0	0	0	0	0	0	0	59
TOTAL:	4,972.5	4,972.5	0	0	0	0	0	0	22,324.4	22,324.4	376

* Shells with attached nozzles have weight reduced by material cut out for opening.

Component	Weight (lb) Contributed by Attachments									Surface Area ft²
	Body Flanges		Nozzles & Flanges		Packed Beds	Trays	Tray Supports	Rings & Clips	Vertical Loads	
	New	Corroded	New	Corroded						
Ellipsoidal Head #1	0	0	198.9	198.9	0	0	0	0	0	1
Cylinder #1	0	0	102.1	102.1	0	0	0	0	0	6
Cylinder #2	0	0	135.9	135.9	0	0	0	0	0	7
Ellipsoidal Head #2	0	0	0	0	0	0	0	0	0	0
TOTAL:	0	0	436.9	436.9	0	0	0	0	0	13

Vessel operating weight, Corroded: 5,409 lb

Vessel operating weight, New: 5,409 lb

Vessel empty weight, Corroded: 5,409 lb

Vessel empty weight, New: 5,409 lb

Vessel test weight, New: 27,734 lb

Vessel test weight, Corroded: 27,734 lb

Vessel surface area: 388 ft²

Vessel center of gravity location - from datum - lift condition

Vessel Lift Weight, New: 5,409 lb

Center of Gravity: 99.2098"

Vessel Capacity

Vessel Capacity** (New): 2,657 US gal

Vessel Capacity** (Corroded): 2,657 US gal

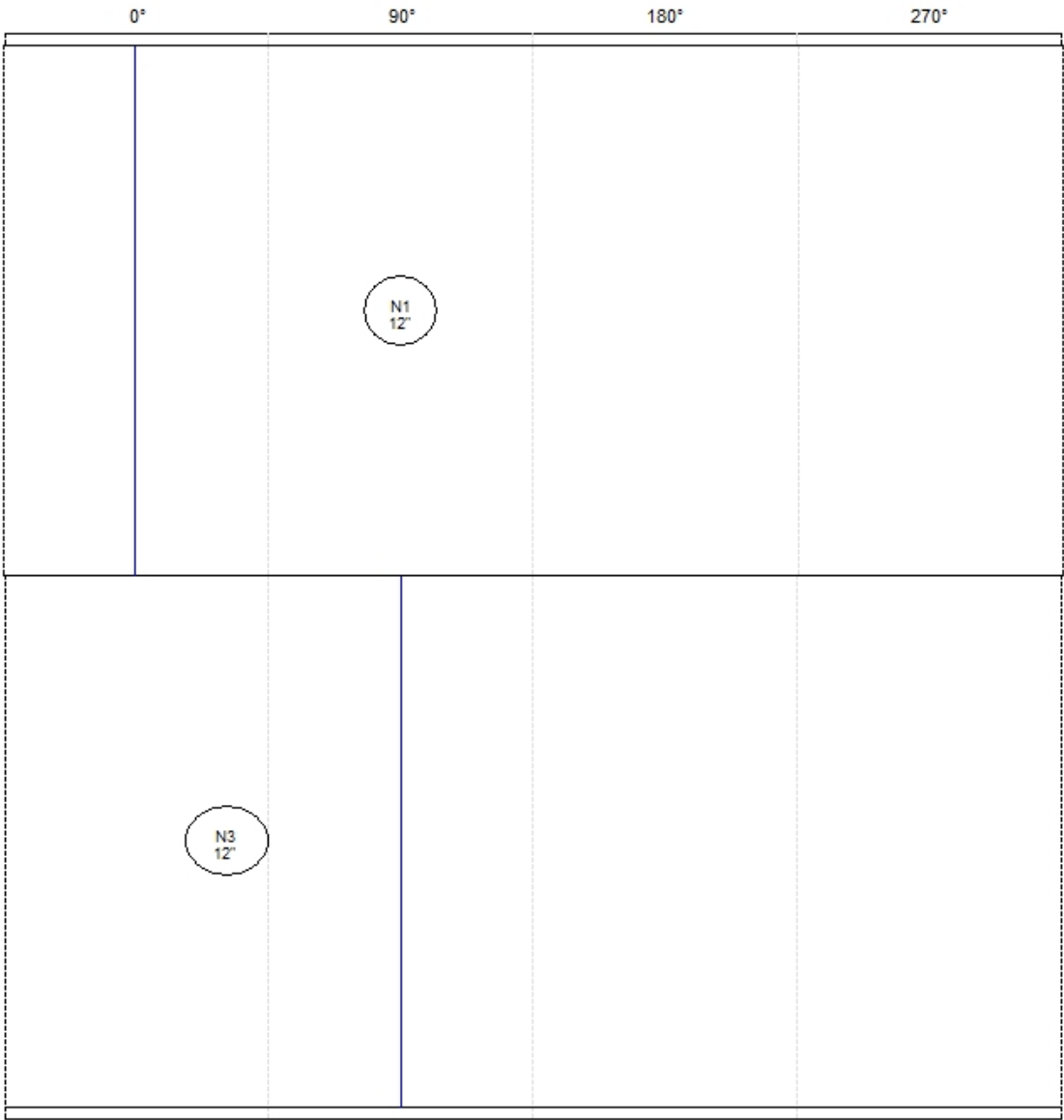
**The vessel capacity does not include volume of nozzle, piping or other attachments.

Long Seam Summary

Shell Long Seam Angles	
Component	Seam 1
Cylinder #1	0°
Cylinder #2	90°

Shell Plate Lengths		
Component	Starting Angle	Plate 1
Cylinder #1	0°	190.4716"
Cylinder #2	90°	189.925"

*Plate Lengths use the circumference of the vessel based on the mid diameter of the components



Shell Rollout

Test Report

The shop test condition has not been investigated for the Chamber bounded by Ellipsoidal Head #2 and Ellipsoidal Head #1.

The field test condition has not been investigated for the Chamber bounded by Ellipsoidal Head #2 and Ellipsoidal Head #1.

Inspection on 12/12/2013

Calculation Results Summary			
Projected Retirement Date:	8/10/2845	Governing Component:	Nozzle #3 (N3)
Governing CML:	CML 28 on Cylinder #2	Projected Thickness at Retirement Date:	0.1621"
Target MAWP:	100 psi @ 650 °F	Target MAEP:	N/A

Equipment Information	
Vessel Name:	7400

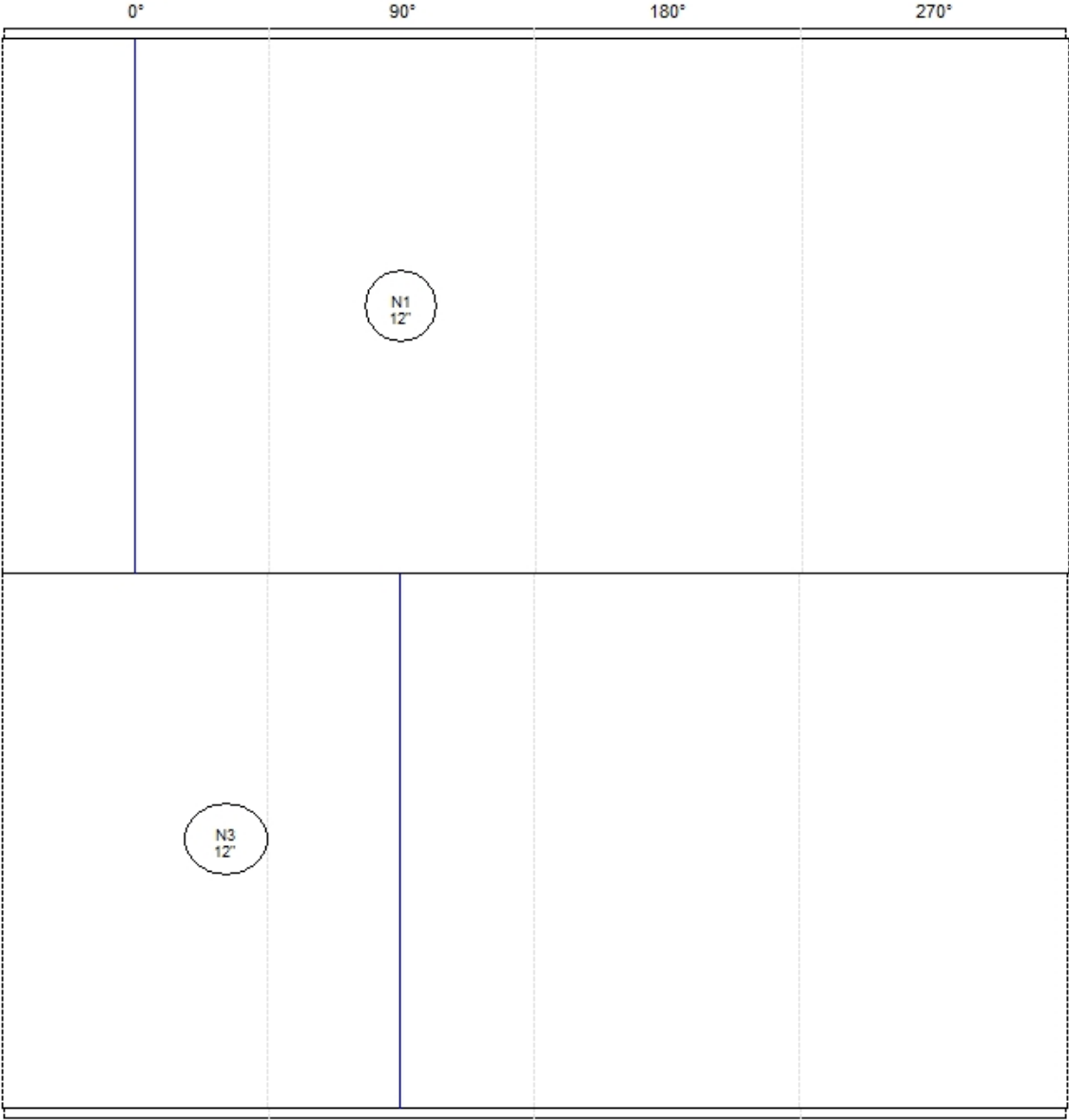
Summary of Thickness Measurements								
Component	Previous Average Thickness	Average Measured Thickness	Governing CML Thickness	Minimum Required Thickness	Short Term Corrosion Rate	Long Term Corrosion Rate	Remaining Life	Governing Component
Ellipsoidal Head #1	0.23"	0.23"	0.23"	0.1597"	0"/yr	0"/yr	-	No
Cylinder #1	0.375"	0.3748"	0.371"	0.1616"	0.0002"/yr	0.0002"/yr	1,047.111 years	No
Cylinder #2	0.375"	0.3748"	0.37"	0.1621"	0.00025"/yr	0.00025"/yr	831.66 years	No
Ellipsoidal Head #2	0.23"	0.23"	0.23"	0.1597"	0"/yr	0"/yr	-	No
Nozzle #3 (N3)	0.375"	0.375"	0.375"	0.375"	0"/yr	0"/yr	831.66 years	Yes

*The Previous Average Thickness is average measured thickness from the previous thickness inspection of each component.

Inspection on 12/12/2013			
Inspection Date:	12/12/2013	Report Date:	12/12/2013
Inspections Performed			
Visual Inspection(VT): External	YES	Visual Inspection(VT): Internal	YES
Thickness Inspection:	YES		

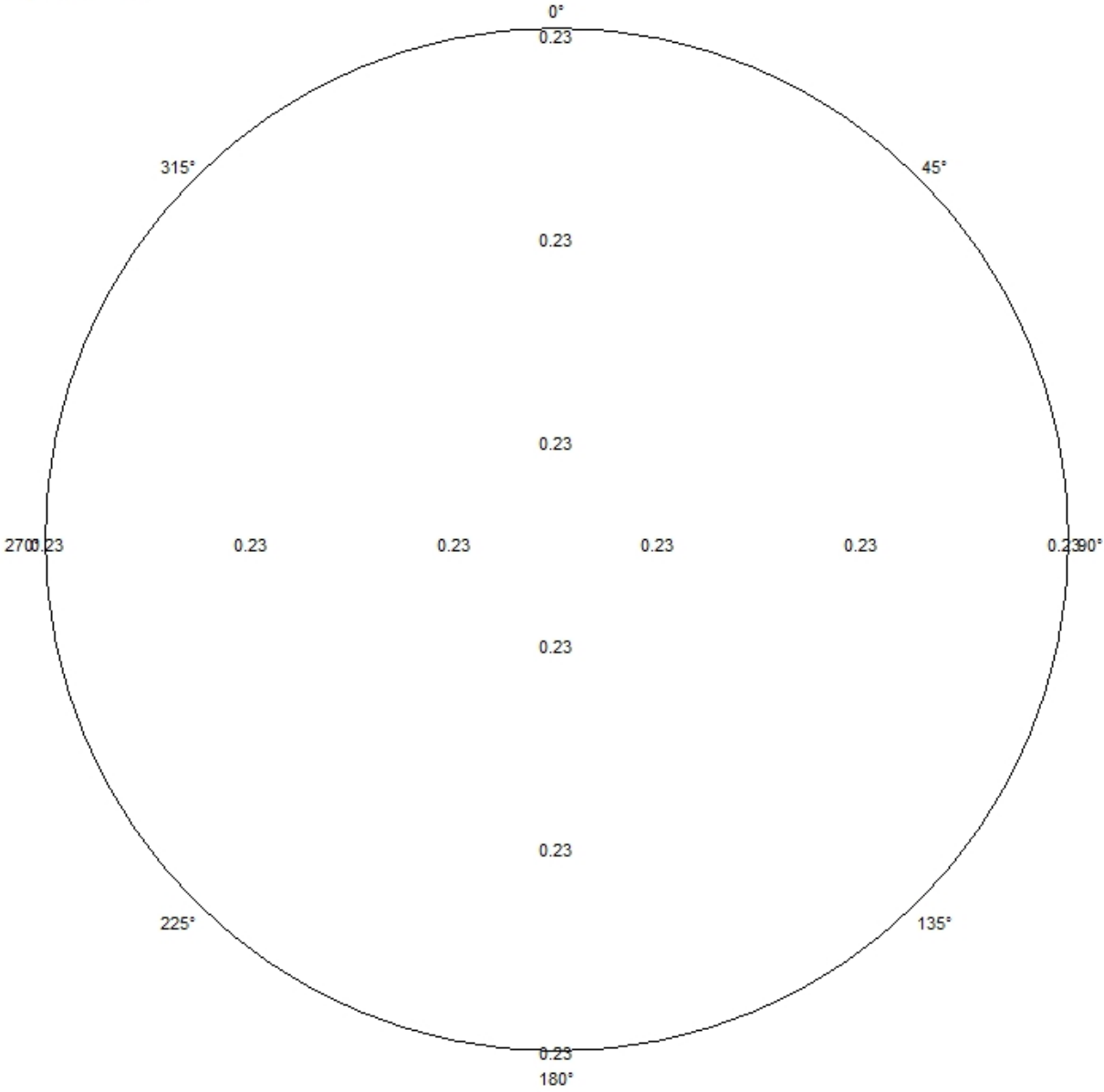
Shell Courses						
Course	ID	Course Length	Allowable Stress	Joint Efficiency (circ, long)	Actual Thickness	Required Thickness
1	Straight Flange on Ellipsoidal Head #1	192" - 194"	18,800 psi	1, 1	N/A	0.1601"
2	Cylinder #1	96" - 192"	18,800 psi	1, 1	0.371"	0.1616"
3	Cylinder #2	0" - 96"	18,800 psi	1, 1	0.37"	0.1621"
4	Straight Flange on Ellipsoidal Head #2	-2" - 0"	18,800 psi	1, 1	N/A	0.1601"

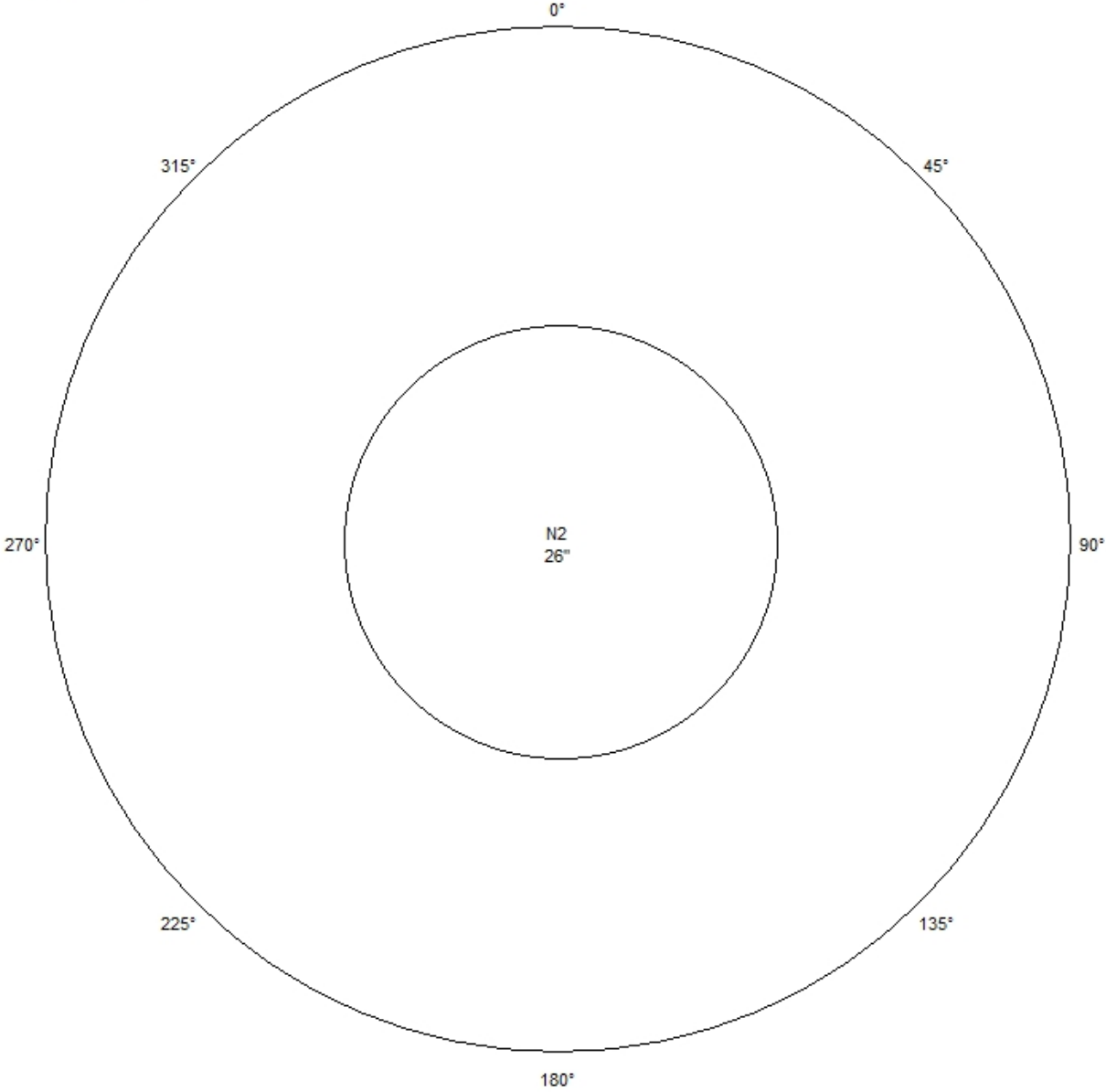
Shell Measured Thickness				
	0 °	90 °	180 °	270 °
182"	0.375"	0.375"	0.375"	0.375"
170"	0.375"	0.375"	0.375"	0.375"
158"	0.375"	0.375"	0.375"	0.375"
146"	0.374"	0.375"	0.371"	0.375"
134"	0.375"	0.375"	0.375"	0.375"
122"	0.375"	0.375"	0.375"	0.375"
110"	0.375"	0.373"	0.375"	0.375"
98"	0.375"	0.375"	0.375"	0.375"
86"	0.375"	0.375"	0.375"	0.375"
74"	0.375"	0.375"	0.37"	0.375"
62"	0.375"	0.375"	0.375"	0.375"
50"	0.375"	0.375"	0.375"	0.375"
38"	0.375"	0.375"	0.375"	0.375"
26"	0.375"	0.375"	0.375"	0.375"
14"	0.375"	0.375"	0.375"	0.375"
2"	0.375"	0.375"	0.375"	0.375"



Ellipsoidal Head #1 Measured Thickness				
	0 °	90 °	180 °	270 °
30"	0.23"	0.23"	0.23"	0.23"
18"	0.23"	0.23"	0.23"	0.23"
6"	0.23"	0.23"	0.23"	0.23"

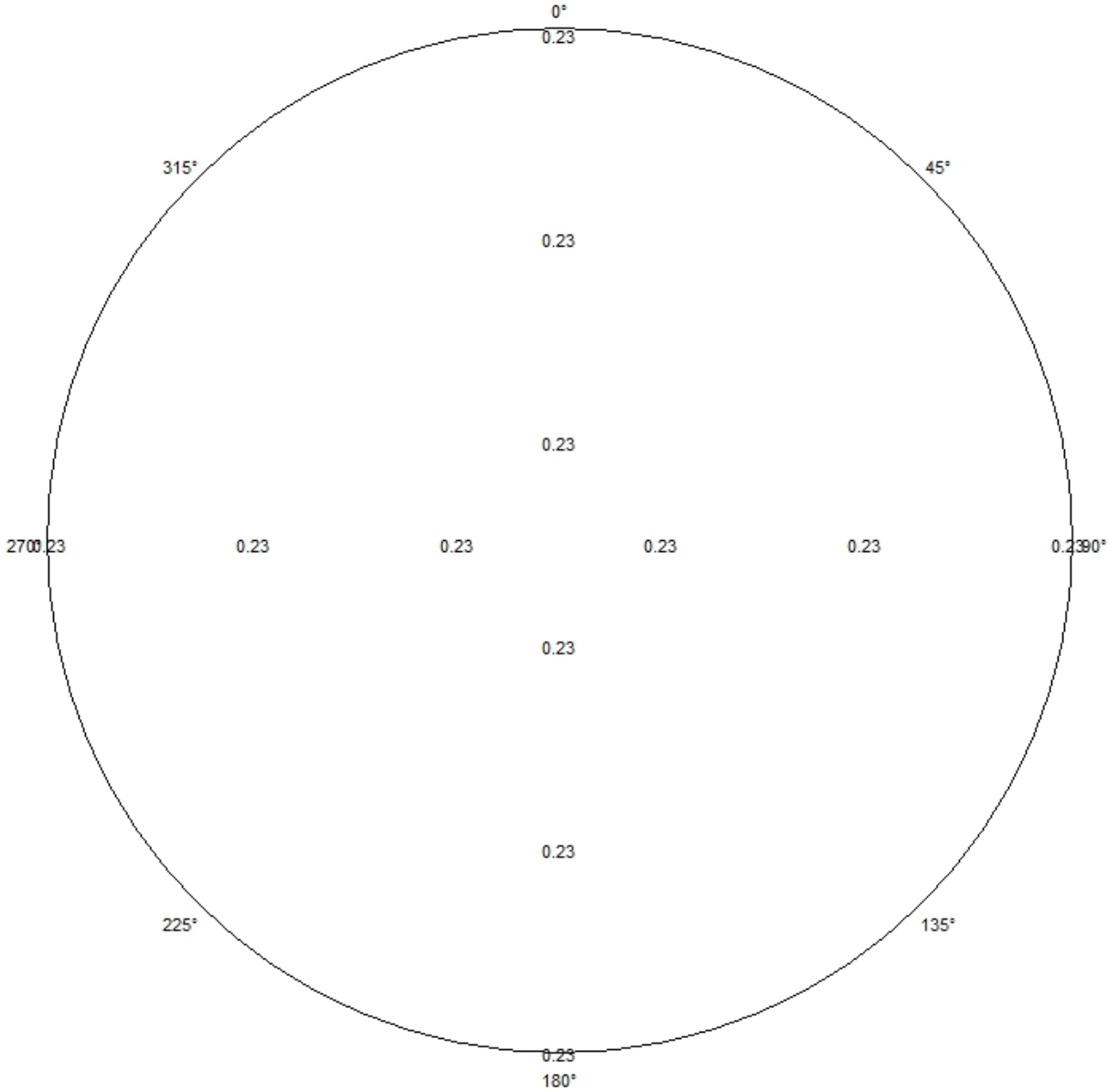
Ellipsoidal Head #1





Ellipsoidal Head #2 Measured Thickness				
	0 °	90 °	180 °	270 °
30"	0.23"	0.23"	0.23"	0.23"
18"	0.23"	0.23"	0.23"	0.23"
6"	0.23"	0.23"	0.23"	0.23"

Ellipsoidal Head #2



Nozzle Readings			
	Dist from Nozzle End	0 °	180 °
N2	1.2191"	4.75"	4.75"
N1	7.9063"	0.375"	0.375"
N3	8.5314"	0.375"	0.375"

Shell Details							
Component	Measured Thickness	Min. Required Thickness	Length	Rem Corr.	Governs	Corr. Allow	Material
Cylinder #1	0.371"	0.1616"	96"	0.2094"	No	0"	SA-516 70
Cylinder #2	0.37"	0.1621"	96"	0.2079"	No	0"	SA-516 70

Head Details						
Component	Measured Thickness	Min. Required Thickness	Rem Corr.	Governs	Corr. Allow	Material
Ellipsoidal Head #1	0.23"	0.1597"	0.0703"	No	0"	SA-516 70
Ellipsoidal Head #2	0.23"	0.1597"	0.0703"	No	0"	SA-516 70

Nozzle Details				
Nozzle	Minimum Measured Thickness	Governs	Corr. Allow	Material
Nozzle #2 (N2)	4.75"	No	0"	SA-105
Nozzle #1 (N1)	0.375"	No	0"	SA-106 B
Nozzle #3 (N3)	0.375"	Yes	0"	SA-106 B

*The minimum required thickness may be calculated for each nozzle by activating the switch on the *Remaining Life Basis* dialog.

Remaining Life Calculation

Governing Component: [Nozzle #3 \(N3\)](#) due to corrosion on Cylinder #2
Corrosion Basis: Short Term Corrosion Rate (ST)
Target MAWP: 100 psi @ 650 °F
Target MAEP: N/A
Current Inspection Date: 12/12/2013

Governing CML

CML ID: CML 28 on Cylinder #2
Minimum Required Thickness, t_{required} : 0.1621"
Measured Thickness, t_{actual} : 0.37"

Measured On: 12/12/2013
Previous Measured Thickness, t_{previous} : 0.375"
Previously Measured On: 12/12/1993
Time Between Measurements: 20 years

*Nozzle #3 (N3) governs, yet does not have a valid corrosion rate.

* t_{required} in this case is the thickness of Cylinder #2, at the point where Nozzle #3 (N3) fails.

Corrosion Rate

$$\begin{aligned}\text{Corrosion Rate} &= (t_{\text{previous}} - t_{\text{actual}}) / \text{Years Difference} \\ &= (0.375 - 0.37) / 20 \\ &= 0.00025"/\text{yr}\end{aligned}$$

Remaining Life

$$\begin{aligned}\text{Remaining Life} &= (t_{\text{actual}} - t_{\text{required}}) / \text{Corrosion Rate} \\ &= (0.37 - 0.1621) / 0.00025 \\ &= 831.66 \text{ years}\end{aligned}$$

Thickness Inspection Interval, API 510 6.5

415.83 years or 10 years (whichever is less) = 10 years
Next Thickness Inspection required by 12/12/2023

Internal Inspection Interval, API 510 6.5

415.83 years or 10 years (whichever is less) = 10 years
Next Internal Inspection required by 12/12/2023

External Inspection Interval, API 510 6.4

415.83 years or 5 years (whichever is less) = 5 years
Next External Inspection required by 12/12/2018

Projected Retirement Date

12/12/2013 + 831.66 years = 8/10/2845

Cylinder #1

ASME Section VIII Division 1, 2013 Edition (Code Case 2695)				
Component		Cylinder		
Material		SA-516 70 (II-D p. 22, ln. 6)		
Impact Tested	Normalized	Fine Grain Practice	PWHT	Optimize MDMT/ Find MAWP
No	No	No	No	No
		Design Pressure (psi)	Design Temperature (°F)	Design MDMT (°F)
Internal		100	650	-20
Dimensions				
Inner Diameter		60.258"		
Length		96"		
Nominal Thickness		0.371"		
Corrosion	Inner	0"		
	Outer	0"		
		Weight (lb)		Capacity (US gal)
New		1,906.42		1,185.16
Corroded		1,906.42		1,185.16
Radiography				
Longitudinal seam		Full UW-11(a) Type 1		
Left Circumferential seam		Full UW-11(a) Type 1		
Right Circumferential seam		Full UW-11(a) Type 1		

Results Summary	
Governing condition	Operating Hot & Corroded
Design thickness due to internal pressure (t)	0.1607"
Minimum thickness per 4.1.2	0.0625"
Maximum allowable working pressure (MAWP)	230.08 psi

Division 2 4.3.3.1 Design Thickness for Internal Pressure	
$t_d = D / 2 * (\exp[P / (S * E)] - 1) + \text{Corrosion}$	(4.3.1)
Operating Hot & Corroded	
$t_d = 60.258 / 2 * (\exp[(100 + 0) / (18,800 * 1)] - 1) + 0 =$	0.1607"
Operating Cold & Corroded	
$t_r = 60.258 / 2 * (\exp[(144.02 + 0) / (20,000 * 1)] - 1) =$	0.2177"

Division 2 4.3.3.1 Maximum Pressure Circumferential Stress	
$P_{\max} = S * E * \ln[D_o / D]$	(4.3.1)
MAWP	
$P_{\max} = 18,800 * 1 * \ln[61 / 60.258] - 0 =$	230.08 psi

Cylinder #2

ASME Section VIII Division 1, 2013 Edition (Code Case 2695)				
Component		Cylinder		
Material		SA-516 70 (II-D p. 22, ln. 6)		
Impact Tested	Normalized	Fine Grain Practice	PWHT	Optimize MDMT/ Find MAWP
No	No	No	No	No
		Design Pressure (psi)	Design Temperature (°F)	Design MDMT (°F)
Internal		100	650	-20
Dimensions				
Inner Diameter		60.085"		
Length		96"		
Nominal Thickness		0.37"		
Corrosion	Inner	0"		
	Outer	0"		
		Weight (lb)		Capacity (US gal)
New		1,895.79		1,178.37
Corroded		1,895.79		1,178.37
Radiography				
Longitudinal seam		Full UW-11(a) Type 1		
Left Circumferential seam		Full UW-11(a) Type 1		
Right Circumferential seam		Full UW-11(a) Type 1		

Results Summary	
Governing condition	Operating Hot & Corroded
Design thickness due to internal pressure (t)	0.1602"
Minimum thickness per 4.1.2	0.0625"
Maximum allowable working pressure (MAWP)	230.12 psi

Division 2 4.3.3.1 Design Thickness for Internal Pressure	
$t_d = D / 2 * (\exp[P / (S * E)] - 1) + \text{Corrosion}$	(4.3.1)
Operating Hot & Corroded	
$t_d = 60.085 / 2 * (\exp[(100 + 0) / (18,800 * 1)] - 1) + 0 =$	0.1602"
Operating Cold & Corroded	
$t_r = 60.085 / 2 * (\exp[(144.02 + 0) / (20,000 * 1)] - 1) =$	0.2171"

Division 2 4.3.3.1 Maximum Pressure Circumferential Stress	
$P_{\max} = S * E * \ln[D_o / D]$	(4.3.1)
MAWP	
$P_{\max} = 18,800 * 1 * \ln[60.825 / 60.085] - 0 =$	230.12 psi

Ellipsoidal Head #1

ASME Section VIII Division 1, 2013 Edition

Component: Ellipsoidal Head
Material Specification: SA-516 70 (II-D p.22, ln. 6)
Internal design pressure: $P = 100 \text{ psi @ } 650 \text{ }^{\circ}\text{F}$

Static liquid head:

$P_s = 0 \text{ psi}$ ($SG=1$, $H_s=0$ " Operating head)

Corrosion allowance: Inner $C = 0$ " Outer $C = 0$ "

Design MDMT = -20°F No impact test performed
Rated MDMT = N/A Material is not normalized
Material is not produced to fine grain practice
PWHT is not performed
Do not Optimize MDMT / Find MAWP

Radiography: Category A joints - Seamless No RT
Head to shell seam - Full UW-11(a) Type 1

Estimated weight*: new = 279.5 lb corr = 279.5 lb
Capacity*: new = 146.9 US gal corr = 146.9 US gal
* includes straight flange

Inner diameter = 60"
Minimum head thickness = 0.23"
Head ratio $D/2h$ = 2 (new)
Head ratio $D/2h$ = 2 (corroded)
Straight flange length L_{sf} = 2"
Nominal straight flange thickness t_{sf} = 0.375"

Results Summary

The governing condition is internal pressure.
Minimum thickness per UG-16 = $0.0625" + 0" = 0.0625"$
Design thickness due to internal pressure (t) = [0.1597"](#)
Maximum allowable working pressure (MAWP) = [144.02](#) psi
Maximum allowable external pressure (MAEP) = [28.01](#) psi

Design thickness for internal pressure, (Corroded at 650 °F) UG-32(d)(1)

$$\begin{aligned} t &= P \cdot D / (2 \cdot S \cdot E - 0.2 \cdot P) + \text{Corrosion} \\ &= 100 \cdot 60 / (2 \cdot 18,800 \cdot 1 - 0.2 \cdot 100) + 0 \\ &= 0.1597" \end{aligned}$$

The head internal pressure design thickness is [0.1597"](#).

Maximum allowable working pressure, (Corroded at 650 °F) UG-32(d)(1)

$$\begin{aligned} P &= 2 \cdot S \cdot E \cdot t / (D + 0.2 \cdot t) - P_s \\ &= 2 \cdot 18,800 \cdot 1 \cdot 0.23 / (60 + 0.2 \cdot 0.23) - 0 \\ &= 144.02 \text{ psi} \end{aligned}$$

The maximum allowable working pressure (MAWP) is [144.02](#) psi.

Maximum Allowable External Pressure, (Corroded at 650 °F) UG-33(d)

Equivalent outside spherical radius (R_o)

$$\begin{aligned} R_o &= K_o \cdot D_o \\ &= 0.8932 \cdot 60.46 \\ &= 54.0031 \text{ in} \end{aligned}$$

$$\begin{aligned} A &= 0.125 / (R_o / t) \\ &= 0.125 / (54.0031 / 0.23) \\ &= 0.000532 \end{aligned}$$

From Table CS-2: $B = 6,576.1909$ psi

$$\begin{aligned} P_a &= B / (R_o / t) \\ &= 6,576.1909 / (54.0031 / 0.23) \\ &= 28.0081 \text{ psi} \end{aligned}$$

Check the Maximum External Pressure, UG-33(a)(1) UG-32(d)(1)

$$\begin{aligned} P &= 2 \cdot S \cdot E \cdot t / ((D + 0.2 \cdot t) \cdot 1.67) - P_{s2} \\ &= 2 \cdot 18,800 \cdot 1 \cdot 0.23 / ((60 + 0.2 \cdot 0.23) \cdot 1.67) - 0 \\ &= 86.24 \text{ psi} \end{aligned}$$

The maximum allowable external pressure (MAEP) is [28.01](#) psi.

29/98

The extreme fiber elongation does not exceed 5%.

Allowable Compressive Stress, Hot and Corroded- S_{cHC} , (table CS-2)

$$\begin{aligned} A &= 0.125 / (R_o / t) \\ &= 0.125 / (30.375 / 0.375) \\ &= 0.001543 \\ B &= 9,261 \text{ psi} \\ S &= 18,800 / 1.00 = 18,800 \text{ psi} \\ S_{cHC} &= \min(B, S) = 9,261 \text{ psi} \end{aligned}$$

Allowable Compressive Stress, Hot and New- S_{cHN}

$$\begin{aligned} S_{cHN} &= S_{cHC} \\ &= 9,261 \text{ psi} \end{aligned}$$

Allowable Compressive Stress, Cold and New- S_{cCN} , (table CS-2)

$$\begin{aligned} A &= 0.125 / (R_o / t) \\ &= 0.125 / (30.375 / 0.375) \\ &= 0.001543 \\ B &= 13,927 \text{ psi} \\ S &= 20,000 / 1.00 = 20,000 \text{ psi} \\ S_{cCN} &= \min(B, S) = 13,927 \text{ psi} \end{aligned}$$

Allowable Compressive Stress, Cold and Corroded- S_{cCC}

$$\begin{aligned} S_{cCC} &= S_{cCN} \\ &= 13,927 \text{ psi} \end{aligned}$$

Allowable Compressive Stress, Vacuum and Corroded- S_{cVC} , (table CS-2)

$$\begin{aligned} A &= 0.125 / (R_o / t) \\ &= 0.125 / (30.375 / 0.375) \\ &= 0.001543 \\ B &= 9,261 \text{ psi} \\ S &= 18,800 / 1.00 = 18,800 \text{ psi} \\ S_{cVC} &= \min(B, S) = 9,261 \text{ psi} \end{aligned}$$

Ellipsoidal Head #1 (Dished Region)

ASME Section VIII Division 1, 2013 Edition

Component: Hemi Head
Material specification: SA-516 70 (II-D p. 22, ln. 6)

Internal design pressure: $P = 100 \text{ psi @ } 650 \text{ }^{\circ}\text{F}$

Corrosion allowance Inner $C = 0''$ Outer $C = 0''$

PWHT is not performed

Radiography: RT on Category A Joints - Long. Seam - Seamless No RT
 RT on Category A Joints - Circ. Seam - User Defined ($E = 1.$)

Estimated weight New = 1,187.6 lb corr = 1,187.6 lb
Capacity New = 1,409.8 US gal corr = 1,409.8 US gal

Inner diameter = 107.5463"

Minimum head
thickness = 0.23"

Design thickness, (at 650 °F) UG-32(f)

$$\begin{aligned} t &= P \cdot R / (2 \cdot S \cdot E - 0.20 \cdot P) + \text{Corrosion} \\ &= 100 \cdot 53.7731 / (2 \cdot 18,800 \cdot 1.00 - 0.20 \cdot 100) + 0 \\ &= 0.1431'' \end{aligned}$$

Maximum allowable working pressure, (at 650 °F) UG-32(f)

$$\begin{aligned} P &= 2 \cdot S \cdot E \cdot t / (R + 0.20 \cdot t) - P_s \\ &= 2 \cdot 18,800 \cdot 1.00 \cdot 0.23 / (53.7731 + 0.20 \cdot 0.23) - 0 \\ &= 160.69 \text{ psi} \end{aligned}$$

Maximum Allowable External Pressure, (Corroded @ 650 °F)

$$\begin{aligned} A &= 0.125 / (R_o / t) \\ &= 0.125 / (54.0031 / 0.23) \\ &= 0.000532 \\ \text{From table CS-2: } B &= 6,576.1909 \text{ psi} \end{aligned}$$

$$\begin{aligned} P_a &= B / (R_o / t) \\ &= 6,576.19 / (54.0031 / 0.23) \\ &= 28.01 \text{ psi} \end{aligned}$$

The maximum allowable external pressure is 28.01 psi

% Extreme fiber elongation - UCS-79(d)

$$\begin{aligned} \text{EFE} &= (75 \cdot t / R_f) \cdot (1 - R_f / R_o) \\ &= (75 \cdot 0.23 / 53.8881) \cdot (1 - 53.8881 / \infty) \\ &= 0.3201\% \end{aligned}$$

The extreme fiber elongation does not exceed 5%.

Head design thickness = 0.1431"

The governing condition is due to internal pressure.

The head thickness of 0.23" is adequate.

The extreme fiber elongation does not exceed 5%.

Allowable Compressive Stress, Hot and Corroded- S_{cHC} , (table CS-2)

$$\begin{aligned} A &= 0.125 / (R_o / t) \\ &= 0.125 / (30.375 / 0.375) \\ &= 0.001543 \\ B &= 9,261 \text{ psi} \\ S &= 18,800 / 1.00 = 18,800 \text{ psi} \\ S_{cHC} &= \min(B, S) = 9,261 \text{ psi} \end{aligned}$$

Allowable Compressive Stress, Hot and New- S_{cHN}

$$\begin{aligned} S_{cHN} &= S_{cHC} \\ &= 9,261 \text{ psi} \end{aligned}$$

Allowable Compressive Stress, Cold and New- S_{cCN} , (table CS-2)

$$\begin{aligned} A &= 0.125 / (R_o / t) \\ &= 0.125 / (30.375 / 0.375) \\ &= 0.001543 \\ B &= 13,927 \text{ psi} \\ S &= 20,000 / 1.00 = 20,000 \text{ psi} \\ S_{cCN} &= \min(B, S) = 13,927 \text{ psi} \end{aligned}$$

Allowable Compressive Stress, Cold and Corroded- S_{cCC}

$$\begin{aligned} S_{cCC} &= S_{cCN} \\ &= 13,927 \text{ psi} \end{aligned}$$

Allowable Compressive Stress, Vacuum and Corroded- S_{cVC} , (table CS-2)

$$\begin{aligned} A &= 0.125 / (R_o / t) \\ &= 0.125 / (30.375 / 0.375) \\ &= 0.001543 \\ B &= 9,261 \text{ psi} \\ S &= 18,800 / 1.00 = 18,800 \text{ psi} \\ S_{cVC} &= \min(B, S) = 9,261 \text{ psi} \end{aligned}$$

Ellipsoidal Head #2

ASME Section VIII Division 1, 2013 Edition

Component: Ellipsoidal Head
Material Specification: SA-516 70 (II-D p.22, ln. 6)
Internal design pressure: $P = 100 \text{ psi @ } 650 \text{ }^{\circ}\text{F}$

Static liquid head:

$P_s = 0 \text{ psi}$ ($SG=1$, $H_s=0$ " Operating head)

Corrosion allowance: Inner $C = 0$ " Outer $C = 0$ "

Design MDMT = -20°F No impact test performed
Rated MDMT = N/A Material is not normalized
Material is not produced to fine grain practice
PWHT is not performed
Do not Optimize MDMT / Find MAWP

Radiography: Category A joints - Seamless No RT
Head to shell seam - Full UW-11(a) Type 1

Estimated weight*: new = 312.8 lb corr = 312.8 lb
Capacity*: new = 146.9 US gal corr = 146.9 US gal
* includes straight flange

Inner diameter = 60"
Minimum head thickness = 0.23"
Head ratio $D/2h$ = 2 (new)
Head ratio $D/2h$ = 2 (corroded)
Straight flange length L_{sf} = 2"
Nominal straight flange thickness t_{sf} = 0.375"

Results Summary

The governing condition is internal pressure.
Minimum thickness per UG-16 = $0.0625" + 0" = 0.0625"$
Design thickness due to internal pressure (t) = [0.1597"](#)
Maximum allowable working pressure (MAWP) = [144.02](#) psi
Maximum allowable external pressure (MAEP) = [28.01](#) psi

Design thickness for internal pressure, (Corroded at 650 °F) UG-32(d)(1)

$$\begin{aligned} t &= P \cdot D / (2 \cdot S \cdot E - 0.2 \cdot P) + \text{Corrosion} \\ &= 100 \cdot 60 / (2 \cdot 18,800 \cdot 1 - 0.2 \cdot 100) + 0 \\ &= 0.1597" \end{aligned}$$

The head internal pressure design thickness is [0.1597"](#).

Maximum allowable working pressure, (Corroded at 650 °F) UG-32(d)(1)

$$\begin{aligned} P &= 2 \cdot S \cdot E \cdot t / (D + 0.2 \cdot t) - P_s \\ &= 2 \cdot 18,800 \cdot 1 \cdot 0.23 / (60 + 0.2 \cdot 0.23) - 0 \\ &= 144.02 \text{ psi} \end{aligned}$$

The maximum allowable working pressure (MAWP) is [144.02](#) psi.

Maximum Allowable External Pressure, (Corroded at 650 °F) UG-33(d)

Equivalent outside spherical radius (R_o)

$$\begin{aligned} R_o &= K_o \cdot D_o \\ &= 0.8932 \cdot 60.46 \\ &= 54.0031 \text{ in} \end{aligned}$$

$$\begin{aligned} A &= 0.125 / (R_o / t) \\ &= 0.125 / (54.0031 / 0.23) \\ &= 0.000532 \end{aligned}$$

From Table CS-2: $B = 6,576.1909$ psi

$$\begin{aligned} P_a &= B / (R_o / t) \\ &= 6,576.1909 / (54.0031 / 0.23) \\ &= 28.0081 \text{ psi} \end{aligned}$$

Check the Maximum External Pressure, UG-33(a)(1) UG-32(d)(1)

$$\begin{aligned} P &= 2 \cdot S \cdot E \cdot t / ((D + 0.2 \cdot t) \cdot 1.67) - P_{s2} \\ &= 2 \cdot 18,800 \cdot 1 \cdot 0.23 / ((60 + 0.2 \cdot 0.23) \cdot 1.67) - 0 \\ &= 86.24 \text{ psi} \end{aligned}$$

The maximum allowable external pressure (MAEP) is [28.01](#) psi.

Ellipsoidal Head #2 (Dished Region)

ASME Section VIII Division 1, 2013 Edition

Component: Hemi Head
Material specification: SA-516 70 (II-D p. 22, ln. 6)

Internal design pressure: $P = 100 \text{ psi @ } 650 \text{ }^{\circ}\text{F}$

Corrosion allowance Inner $C = 0''$ Outer $C = 0''$

PWHT is not performed

Radiography: RT on Category A Joints - Long. Seam - Seamless No RT
 RT on Category A Joints - Circ. Seam - User Defined ($E = 1.$)

Estimated weight New = 1,187.6 lb corr = 1,187.6 lb
Capacity New = 1,409.8 US gal corr = 1,409.8 US gal

Inner diameter = 107.5463"

Minimum head
thickness = 0.23"

Design thickness, (at 650 °F) UG-32(f)

$$\begin{aligned} t &= P \cdot R / (2 \cdot S \cdot E - 0.20 \cdot P) + \text{Corrosion} \\ &= 100 \cdot 53.7731 / (2 \cdot 18,800 \cdot 1.00 - 0.20 \cdot 100) + 0 \\ &= 0.1431'' \end{aligned}$$

Maximum allowable working pressure, (at 650 °F) UG-32(f)

$$\begin{aligned} P &= 2 \cdot S \cdot E \cdot t / (R + 0.20 \cdot t) - P_s \\ &= 2 \cdot 18,800 \cdot 1.00 \cdot 0.23 / (53.7731 + 0.20 \cdot 0.23) - 0 \\ &= 160.69 \text{ psi} \end{aligned}$$

Maximum Allowable External Pressure, (Corroded @ 650 °F)

$$\begin{aligned} A &= 0.125 / (R_o / t) \\ &= 0.125 / (54.0031 / 0.23) \\ &= 0.000532 \\ \text{From table CS-2: } B &= 6,576.1909 \text{ psi} \end{aligned}$$

$$\begin{aligned} P_a &= B / (R_o / t) \\ &= 6,576.19 / (54.0031 / 0.23) \\ &= 28.01 \text{ psi} \end{aligned}$$

The maximum allowable external pressure is 28.01 psi

% Extreme fiber elongation - UCS-79(d)

$$\begin{aligned} \text{EFE} &= (75 \cdot t / R_f) \cdot (1 - R_f / R_o) \\ &= (75 \cdot 0.23 / 53.8881) \cdot (1 - 53.8881 / \infty) \\ &= 0.3201\% \end{aligned}$$

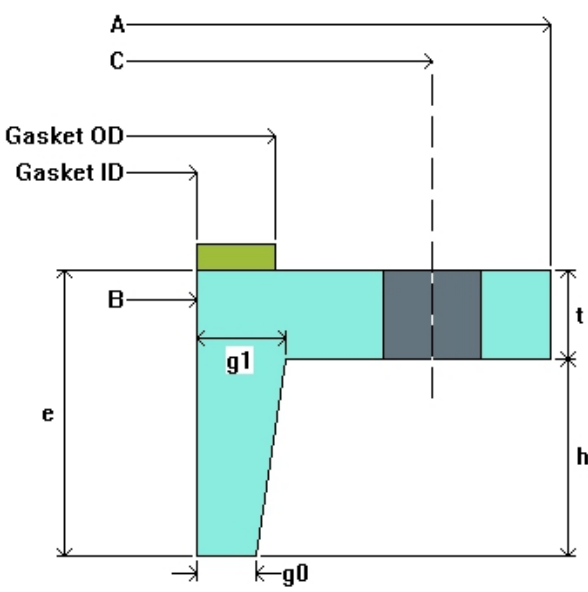
The extreme fiber elongation does not exceed 5%.

Head design thickness = 0.1431"

The governing condition is due to internal pressure.

The head thickness of 0.23" is adequate.

Flange #2

ASME Section VIII Division 1, 2013 Edition (Code Case 2695)				
Flange Type		Weld neck integral		
Flange Material		SA-516 70 (II-D p. 22, ln. 6)		
Attached To		Nozzle #1 (N1)		
Impact Tested	Normalized	Fine Grain Practice	PWHT	Optimize MDMT/ Find MAWP
No	No	No	No	No
		Design Pressure (psi)	Design Temperature (°F)	Design MDMT (°F)
Internal		100	650	-20
Static Liquid Head				
Condition		P_s (psi)		
Operating		0		
Dimensions				
Flange OD, A		16.5"		
Flange ID, B		12"		
Bolt Circle, C		15"		
Gasket OD		13"		
Gasket ID		12"		
Flange Thickness, t		0.5625"		
Hub Thickness, g₁		0.5625"		
Hub Thickness, g₀		0.375"		
Hub Length, h		1.25"		
Length, e		1.8125"		
Bore		0"		
Corrosion				
Flange		0"		
Bolting				
Material		SA-193 B7 Bolt <= 2 1/2 (II-D p. 352, ln. 31)		
Description		8- 0.625 in dia series 8 threaded		
Corrosion on root		0"		
Gasket				
Factor, m		2		
Seating Stress, y		2,500 psi		

Gasket Thickness	0.175"
Weight (lb)	
New	22.5 lb
Corroded	22.5 lb

Results Summary	
Flange design thickness:	0.5525"
Maximum allowable working pressure, MAWP:	166.57 psi @ 650 °F
Rated MDMT	N/A °F

Note: this flange is calculated as an integral type.

Flange calculations for MAWP + Weight Only

Division 2 4.16.6.1 Step 3 - Gasket Factors	
$b_0 = N / 2$	(Table 4.16.3)
$b = b_0$ for $b_0 \leq 0.25$ in	(4.16.1)
$G = (G_c + \text{Gasket ID}) / 2$	(4.16.3)
Results	
Table 4.16.3 Facing Sketch Detail	1(a) or (b), Column II
Gasket width, N	0.5"
$b_0 = 0.5 / 2$	0.25"
$b = 0.25$	0.25"
$G = (13 + 12) / 2$	12.5"

Division 2 4.16.6.1 Step 4 - Design Bolt Load for Operating	
$W_o = 0.785 \cdot G^2 \cdot P + 2 \cdot b \cdot \pi \cdot G \cdot m \cdot P$	(4.16.4)
Results	
$W_o = 0.785 \cdot 12.5^2 \cdot 166.57 + 2 \cdot 0.25 \cdot \pi \cdot 12.5 \cdot 2 \cdot 166.57$	26,972.17 lb _f

Division 2 4.16.6.1 Step 5 - Design Bolt Load for Gasket Seating and Required Bolt Area	
$W_{gs} = \pi \cdot b \cdot G \cdot (C_{us} \cdot y)$	(4.16.8)
$A_m = \max [(W_o + F_A + 4 \cdot M_E / G) / S_{bo} , W_{gs} / S_{bg}]$	(4.16.7)
$W_g = \{ (A_m + A_b) / 2 \} \cdot S_{bg}$	(4.16.6)
Results	
$W_{gs} = \pi \cdot 0.25 \cdot 12.5 \cdot (1 \cdot 2,500)$	24,543.69 lb _f
$A_m = \max [(26,972.17 + 0 + 4 \cdot 0 / 12.5) / 25,000 , 24,543.69 / 25,000]$	1.0789 in ²
Total corroded bolt area for 8- 0.625 in dia bolts, A_b	1.616 in ²
$W_g = \{ (1.0789 + 1.616) / 2 \} \cdot 25,000$	33,686.09 lb _f

Division 2 4.16.7.2 Step 4 - Flange Stress Factors	
$K = A / B$	
$Y = 1 / (K - 1) [0.66845 + 5.7169 \cdot K^2 \cdot \log_{10}[K] / (K^2 - 1)]$	
$T = \{ K^2 \cdot (1 + 8.55246 \cdot \log_{10}[K]) - 1 \} / \{ (1.0472 + 1.9448 \cdot K^2) \cdot (K - 1) \}$	
$U = \{ K^2 \cdot (1 + 8.55246 \cdot \log_{10}[K]) - 1 \} / \{ 1.36136 \cdot (K^2 - 1) \cdot (K - 1) \}$	
$Z = (K^2 + 1) / (K^2 - 1)$	
$h_o = (B \cdot g_o)^{0.5}$	
$X_g = g_1 / g_o$	
$X_h = h / h_o$	
$F = 0.897697 - 0.297012 \cdot \ln[X_g] + 0.0095257 \cdot \ln[X_h] + 0.123586 \cdot (\ln[X_g])^2 + 0.035858 \cdot (\ln[X_h])^2 - 0.194422 \cdot \ln[X_g] \cdot \ln[X_h] \\ - 0.0181259 \cdot (\ln[X_g])^3 + 0.012936 \cdot (\ln[X_h])^3 - 0.0377693 \cdot \ln[X_g] \cdot (\ln[X_h])^2 + 0.0273791 \cdot (\ln[X_g])^2 \cdot \ln[X_h]$	
$V = 0.0144868 - 0.135977 / X_g - 0.0461919 / X_h + 0.560718 / X_g^2 + 0.0529829 / X_h^2 + 0.244313 / (X_g \cdot X_h) + 0.113929 / X_g^3 \\ - 0.00928265 / X_h^3 - 0.0266293 / (X_g \cdot X_h^2) - 0.217008 / (X_g^2 \cdot X_h)$	
$f = \max [1 , (0.0927779 - 0.0336633 \cdot X_g + 0.964176 \cdot X_g^2 + 0.0566286 \cdot X_h + 0.347074 \cdot X_h^2 - 4.18699 \cdot X_h^3) / (1 - 0.00596093 \cdot X_g \\ + 1.62904 \cdot X_h + 3.49329 \cdot X_h^2 + 1.39052 \cdot X_h^3)]$	
$e = F / h_o$	
$d = U \cdot g_o^2 \cdot h_o / V$	
$L = (t \cdot e + 1) / T + t^3 / d$	
Results	
$K = 16.5 / 12$	1.375
$Y = 1 / (1.375 - 1) [0.66845 + 5.7169 \cdot 1.375^2 \cdot \log_{10}[1.375] / (1.375^2 - 1)]$	6.2583
$T = \{ 1.375^2 \cdot (1 + 8.55246 \cdot \log_{10}[1.375]) - 1 \} / \{ (1.0472 + 1.9448 \cdot 1.375^2) \cdot (1.375 - 1) \}$	1.7651
$U = \{ 1.375^2 \cdot (1 + 8.55246 \cdot \log_{10}[1.375]) - 1 \} / \{ 1.36136 \cdot (1.375^2 - 1) \cdot (1.375 - 1) \}$	6.8773
$Z = (1.375^2 + 1) / (1.375^2 - 1)$	3.2456
$h_o = (12 \cdot 0.375)^{0.5}$	2.1213"
$X_g = 0.5625 / 0.375$	1.5
$X_h = 1.25 / 2.1213$	0.5893
$F = 0.897697 - 0.297012 \cdot \ln[1.5] + 0.0095257 \cdot \ln[0.5893] + 0.123586 \cdot (\ln[1.5])^2 + 0.035858 \cdot (\ln[0.5893])^2 \\ - 0.194422 \cdot \ln[1.5] \cdot \ln[0.5893] - 0.0181259 \cdot (\ln[1.5])^3 + 0.012936 \cdot (\ln[0.5893])^3 - 0.0377693 \cdot \ln[1.5] \cdot (\ln[0.5893])^2 \\ + 0.0273791 \cdot (\ln[1.5])^2 \cdot \ln[0.5893]$	0.8345
$V = 0.0144868 - 0.135977 / 1.5 - 0.0461919 / 0.5893 + 0.560718 / 1.5^2 + 0.0529829 / 0.5893^2 + 0.244313 / (1.5 \cdot 0.5893) \\ + 0.113929 / 1.5^3 - 0.00928265 / 0.5893^3 - 0.0266293 / (1.5 \cdot 0.5893^2) - 0.217008 / (1.5^2 \cdot 0.5893)$	0.2972
$f = \max [1 , (0.0927779 - 0.0336633 \cdot 1.5 + 0.964176 \cdot 1.5^2 + 0.0566286 \cdot 0.5893 + 0.347074 \cdot 0.5893^2 - 4.18699 \cdot 0.5893^3) / (1 - 0.00596093 \cdot 1.5 \\ + 1.62904 \cdot 0.5893 + 3.49329 \cdot 0.5893^2 + 1.39052 \cdot 0.5893^3)]$	1
$e = 0.8345 / 2.1213$	0.3934 1/in
$d = 6.8773 \cdot 0.375^2 \cdot 2.1213 / 0.2972$	6.9022 in ³
$L = (0.5625 \cdot 0.3934 + 1) / 1.7651 + 0.5625^3 / 6.9022$	0.7177

Division 2 4.16.7.2 Step 5 - Flange Forces	
$H_D = 0.785 \cdot B^2 \cdot P$	
$H = 0.785 \cdot G^2 \cdot P$	
$H_T = H - H_D$	
$H_G = W_o - H$	
Results	
$H_D = 0.785 \cdot 12^2 \cdot 166.57$	18,829.16 lb _f
$H = 0.785 \cdot 12.5^2 \cdot 166.57$	20,430.95 lb _f
$H_T = 20,430.95 - 18,829.16$	1,601.79 lb _f
$H_G = 26,972.17 - 20,430.95$	6,541.22 lb _f

Division 2 4.16.7.2 Step 6 - Flange Moments of Inertia	
$I = 0.0874 \cdot L \cdot g_o^2 \cdot h_o \cdot B / V$	
$A_R = 0.5 \cdot (A - B)$	
$G_{avg} = 0.5 \cdot (g_o + g_i)$	
$t \geq G_{avg}$	
$A_A = A_R, \quad B_B = t, \quad C_C = h, \quad D_{DG} = G_{avg}$	
$K_{AB} = (A_A \cdot B_B^3) \cdot [1 / 3 - 0.21 \cdot (B_B / A_A) \cdot (1 - 1 / 12 \cdot \{B_B / A_A\}^4)]$	
$K_{CD} = (C_C \cdot D_{DG}^3) \cdot [1 / 3 - 0.105 \cdot (D_{DG} / C_C) \cdot (1 - 1 / 192 \cdot \{D_{DG} / C_C\}^4)]$	
$I_p = K_{AB} + K_{CD}$	
Results	
$I = 0.0874 \cdot 0.7177 \cdot 0.375^2 \cdot 2.1213 \cdot 12 / 0.2972$	0.7554 in ⁴
$A_R = 0.5 \cdot (16.5 - 12)$	2.25"
$G_{avg} = 0.5 \cdot (0.375 + 0.5625)$	0.4688"
$t = 0.5625 \text{ in} \geq G_{avg} = 0.4688 \text{ in}$	
$A_A = 2.25 \text{ in}, \quad B_B = 0.5625 \text{ in}, \quad C_C = 1.25 \text{ in}, \quad D_{DG} = 0.4688 \text{ in}$	
$K_{AB} = (2.25 \cdot 0.5625^3) \cdot [1 / 3 - 0.21 \cdot (0.5625 / 2.25) \cdot (1 - 1 / 12 \cdot \{0.5625 / 2.25\}^4)]$	0.1125 in ⁴
$K_{CD} = (1.25 \cdot 0.4688^3) \cdot [1 / 3 - 0.105 \cdot (0.4688 / 1.25) \cdot (1 - 1 / 192 \cdot \{0.4688 / 1.25\}^4)]$	0.0378 in ⁴
$I_p = 0.1125 + 0.0378$	0.1503 in ⁴

Division 2 4.16.7.2 Step 6 - Flange Moment Arms	
$h_D = (C - B - g_1) / 2$	
$h_T = 1 / 2 * [(C - B) / 2 + h_G]$	
$h_G = (C - G) / 2$	
Results	
$h_D = (15 - 12 - 0.5625) / 2$	1.2188"
$h_T = 1 / 2 * [(15 - 12) / 2 + 1.25]$	1.375"
$h_G = (15 - 12.5) / 2$	1.25"

Division 2 4.16.7.2 Step 6 - Flange Moment for Operating	
$M_{oe} = 4 * M_E * [I / (0.3846 * I_p + I)] * [h_D / (C - 2 * h_D)] + F_A * h_D$	(4.16.16)
$M_o = \text{abs} [(H_D * h_D + H_T * h_T + H_G * h_G + M_{oe}) * F_S]$	(4.16.14)
Results	
$M_{oe} = 4 * 0 * [0.7554 / (0.3846 * 0.1503 + 0.7554)] * [1.2188 / (15 - 2 * 1.2188)] + 0 * 1.2188$	0 lb _f -in
$M_o = \text{abs} [(18,829.16 * 1.2188 + 1,601.79 * 1.375 + 6,541.22 * 1.25 + 0) * 1]$	33,327 lb _f -in

Division 2 4.16.7.2 Step 7 - Flange Moment for Gasket Seating	
$M_g = W_g * (C - G) * F_S / 2$	(4.16.17)
Results	
$M_g = 33,686.09 * (15 - 12.5) * 1 / 2$	42,107.6 lb _f -in

Division 2 4.16.7.2 Step 8 - Flange Stresses			
Operating		Gasket Seating	
$S_H = f \cdot M_o / (L \cdot g_1^{2 \cdot B})$		$S_H = f \cdot M_g / (L \cdot g_1^{2 \cdot B})$	
$S_R = (1.33 \cdot t \cdot e + 1) \cdot M_o / (L \cdot t^2 \cdot B)$		$S_R = (1.33 \cdot t \cdot e + 1) \cdot M_g / (L \cdot t^2 \cdot B)$	
$S_T = Y \cdot M_o / (t^2 \cdot B) - Z \cdot S_R$		$S_T = Y \cdot M_g / (t^2 \cdot B) - Z \cdot S_R$	
$S_H \leq \min[1.5 \cdot S_{fo}, 2.5 \cdot S_{no}]$		$S_H \leq \min[1.5 \cdot S_{fg}, 2.5 \cdot S_{ng}]$	
$S_H \leq 1.5 \cdot S_{fo}$		$S_H \leq 1.5 \cdot S_{fg}$	
$S_R \leq S_{fo}$		$S_R \leq S_{fg}$	
$S_T \leq S_{fo}$		$S_T \leq S_{fg}$	
$(S_H + S_R) / 2 \leq S_{fo}$		$(S_H + S_R) / 2 \leq S_{fg}$	
$(S_H + S_T) / 2 \leq S_{fo}$		$(S_H + S_T) / 2 \leq S_{fg}$	
$S_{fo} = 18,800 \text{ psi}$		$S_{fg} = 20,000 \text{ psi}$	
$S_{no} = 17,100 \text{ psi}$		$S_{ng} = 17,100 \text{ psi}$	
Operating			
	Stress (psi)	Allowable (psi)	Overstressed
$S_H = 1 \cdot 33,327 / (0.7177 \cdot 0.5625^2 \cdot 12)$	12,230	28,200	OK
$S_R = (1.33 \cdot 0.5625 \cdot 0.3934 + 1) \cdot 33,327 / (0.7177 \cdot 0.5625^2 \cdot 12)$	15,829	18,800	OK
$S_T = 6.2583 \cdot 33,327 / (0.5625^2 \cdot 12) - 3.2456 \cdot 15,829$	3,556	18,800	OK
$(S_H + S_R) / 2 = (12,230 + 15,829) / 2$	14,030	18,800	OK
$(S_H + S_T) / 2 = (12,230 + 3,556) / 2$	7,893	18,800	OK
Gasket Seating			
	Stress (psi)	Allowable (psi)	Overstressed
$S_H = 1 \cdot 42,107.6 / (0.7177 \cdot 0.5625^2 \cdot 12)$	15,452	30,000	OK
$S_R = (1.33 \cdot 0.5625 \cdot 0.3934 + 1) \cdot 42,107.6 / (0.7177 \cdot 0.5625^2 \cdot 12)$	20,000	20,000	OK
$S_T = 6.2583 \cdot 42,107.6 / (0.5625^2 \cdot 12) - 3.2456 \cdot 20,000$	4,493	20,000	OK
$(S_H + S_R) / 2 = (15,452 + 20,000) / 2$	17,726	20,000	OK
$(S_H + S_T) / 2 = (15,452 + 4,493) / 2$	9,973	20,000	OK

Division 2 4.16.7.2 Step 10 - Flange Rigidity			
Operating		Gasket Seating	
$J = 52.14 \cdot V \cdot M_o / (L \cdot E_{y_o} \cdot g_o^2 \cdot K_R \cdot h_o) \leq 1$		$J = 52.14 \cdot V \cdot M_g / (L \cdot E_{y_g} \cdot g_o^2 \cdot K_R \cdot h_o) \leq 1$	
$K_R = 0.3$			
Operating			
$J = 52.14 \cdot 0.2972 \cdot 33,327 / (0.7177 \cdot 26.0E+06 \cdot 0.375^2 \cdot 0.3 \cdot 2.1213)$	0.3093	≤ 1	OK
Gasket Seating			
$J = 52.14 \cdot 0.2972 \cdot 42,107.6 / (0.7177 \cdot 29.4E+06 \cdot 0.375^2 \cdot 0.3 \cdot 2.1213)$	0.3456	≤ 1	OK

Flange #3

ASME Section VIII Division 1, 2013 Edition (Code Case 2695)				
Flange Type		Weld neck integral		
Flange Material		SA-516 70 (II-D p. 22, ln. 6)		
Attached To		Nozzle #3 (N3)		
Impact Tested	Normalized	Fine Grain Practice	PWHT	Optimize MDMT/ Find MAWP
No	No	No	No	No
		Design Pressure (psi)	Design Temperature (°F)	Design MDMT (°F)
Internal		100	650	-20
Static Liquid Head				
Condition		P_s (psi)		
Operating		0		
Dimensions				
Flange OD, A		16.5"		
Flange ID, B		12"		
Bolt Circle, C		15"		
Gasket OD		13"		
Gasket ID		12"		
Flange Thickness, t		0.5625"		
Hub Thickness, g₁		0.5625"		
Hub Thickness, g₀		0.375"		
Hub Length, h		1.25"		
Length, e		1.8125"		
Bore		0"		
Corrosion				
Flange		0"		
Bolting				
Material		SA-193 B7 Bolt <= 2 1/2 (II-D p. 352, ln. 31)		
Description		8- 0.625 in dia series 8 threaded		
Corrosion on root		0"		
Gasket				
Factor, m		2		
Seating Stress, y		2,500 psi		

Gasket Thickness	0.175"
Weight (lb)	
New	22.5 lb
Corroded	22.5 lb

Results Summary	
Flange design thickness:	0.5525"
Maximum allowable working pressure, MAWP:	166.57 psi @ 650 °F
Rated MDMT	N/A °F

Note: this flange is calculated as an integral type.

Flange calculations for MAWP + Weight Only

Division 2 4.16.6.1 Step 3 - Gasket Factors	
$b_0 = N / 2$	(Table 4.16.3)
$b = b_0$ for $b_0 \leq 0.25$ in	(4.16.1)
$G = (G_c + \text{Gasket ID}) / 2$	(4.16.3)
Results	
Table 4.16.3 Facing Sketch Detail	1(a) or (b), Column II
Gasket width, N	0.5"
$b_0 = 0.5 / 2$	0.25"
$b = 0.25$	0.25"
$G = (13 + 12) / 2$	12.5"

Division 2 4.16.6.1 Step 4 - Design Bolt Load for Operating	
$W_o = 0.785 \cdot G^2 \cdot P + 2 \cdot b \cdot \pi \cdot G \cdot m \cdot P$	(4.16.4)
Results	
$W_o = 0.785 \cdot 12.5^2 \cdot 166.57 + 2 \cdot 0.25 \cdot \pi \cdot 12.5 \cdot 2 \cdot 166.57$	26,972.17 lb _f

Division 2 4.16.6.1 Step 5 - Design Bolt Load for Gasket Seating and Required Bolt Area	
$W_{gs} = \pi \cdot b \cdot G \cdot (C_{us} \cdot y)$	(4.16.8)
$A_m = \max [(W_o + F_A + 4 \cdot M_E / G) / S_{bo} , W_{gs} / S_{bg}]$	(4.16.7)
$W_g = \{(A_m + A_b) / 2\} \cdot S_{bg}$	(4.16.6)
Results	
$W_{gs} = \pi \cdot 0.25 \cdot 12.5 \cdot (1 \cdot 2,500)$	24,543.69 lb _f
$A_m = \max [(26,972.17 + 0 + 4 \cdot 0 / 12.5) / 25,000 , 24,543.69 / 25,000]$	1.0789 in ²
Total corroded bolt area for 8- 0.625 in dia bolts, A_b	1.616 in ²
$W_g = \{(1.0789 + 1.616) / 2\} \cdot 25,000$	33,686.09 lb _f

Division 2 4.16.7.2 Step 4 - Flange Stress Factors	
$K = A / B$	
$Y = 1 / (K - 1) [0.66845 + 5.7169 K^2 \log_{10}[K] / (K^2 - 1)]$	
$T = \{ K^2 (1 + 8.55246 \log_{10}[K]) - 1 \} / \{ (1.0472 + 1.9448 K^2) (K - 1) \}$	
$U = \{ K^2 (1 + 8.55246 \log_{10}[K]) - 1 \} / \{ 1.36136 (K^2 - 1) (K - 1) \}$	
$Z = (K^2 + 1) / (K^2 - 1)$	
$h_o = (B g_o)^{0.5}$	
$X_g = g_1 / g_o$	
$X_h = h / h_o$	
$F = 0.897697 - 0.297012 \ln[X_g] + 0.0095257 \ln[X_h] + 0.123586 (\ln[X_g])^2 + 0.035858 (\ln[X_h])^2 - 0.194422 \ln[X_g] \ln[X_h] \\ - 0.0181259 (\ln[X_g])^3 + 0.012936 (\ln[X_h])^3 - 0.0377693 \ln[X_g] (\ln[X_h])^2 + 0.0273791 (\ln[X_g])^2 \ln[X_h]$	
$V = 0.0144868 - 0.135977 / X_g - 0.0461919 / X_h + 0.560718 / X_g^2 + 0.0529829 / X_h^2 + 0.244313 / (X_g X_h) + 0.113929 / X_g^3 \\ - 0.00928265 / X_h^3 - 0.0266293 / (X_g X_h^2) - 0.217008 / (X_g^2 X_h)$	
$f = \max [1, (0.0927779 - 0.0336633 X_g + 0.964176 X_g^2 + 0.0566286 X_h + 0.347074 X_h^2 - 4.18699 X_h^3) / (1 - 0.00596093 X_g \\ + 1.62904 X_h + 3.49329 X_h^2 + 1.39052 X_h^3)]$	
$e = F / h_o$	
$d = U g_o^2 h_o / V$	
$L = (t e + 1) / T + t^3 / d$	
Results	
$K = 16.5 / 12$	1.375
$Y = 1 / (1.375 - 1) [0.66845 + 5.7169 \cdot 1.375^2 \log_{10}[1.375] / (1.375^2 - 1)]$	6.2583
$T = \{ 1.375^2 (1 + 8.55246 \log_{10}[1.375]) - 1 \} / \{ (1.0472 + 1.9448 \cdot 1.375^2) (1.375 - 1) \}$	1.7651
$U = \{ 1.375^2 (1 + 8.55246 \log_{10}[1.375]) - 1 \} / \{ 1.36136 (1.375^2 - 1) (1.375 - 1) \}$	6.8773
$Z = (1.375^2 + 1) / (1.375^2 - 1)$	3.2456
$h_o = (12 \cdot 0.375)^{0.5}$	2.1213"
$X_g = 0.5625 / 0.375$	1.5
$X_h = 1.25 / 2.1213$	0.5893
$F = 0.897697 - 0.297012 \ln[1.5] + 0.0095257 \ln[0.5893] + 0.123586 (\ln[1.5])^2 + 0.035858 (\ln[0.5893])^2 \\ - 0.194422 \ln[1.5] \ln[0.5893] - 0.0181259 (\ln[1.5])^3 + 0.012936 (\ln[0.5893])^3 - 0.0377693 \ln[1.5] (\ln[0.5893])^2 \\ + 0.0273791 (\ln[1.5])^2 \ln[0.5893]$	0.8345
$V = 0.0144868 - 0.135977 / 1.5 - 0.0461919 / 0.5893 + 0.560718 / 1.5^2 + 0.0529829 / 0.5893^2 + 0.244313 / (1.5 \cdot 0.5893) \\ + 0.113929 / 1.5^3 - 0.00928265 / 0.5893^3 - 0.0266293 / (1.5 \cdot 0.5893^2) - 0.217008 / (1.5^2 \cdot 0.5893)$	0.2972
$f = \max [1, (0.0927779 - 0.0336633 \cdot 1.5 + 0.964176 \cdot 1.5^2 + 0.0566286 \cdot 0.5893 + 0.347074 \cdot 0.5893^2 - 4.18699 \cdot 0.5893^3) / (1 - 0.00596093 \cdot 1.5 \\ + 1.62904 \cdot 0.5893 + 3.49329 \cdot 0.5893^2 + 1.39052 \cdot 0.5893^3)]$	1
$e = 0.8345 / 2.1213$	0.3934 1/in
$d = 6.8773 \cdot 0.375^2 \cdot 2.1213 / 0.2972$	6.9022 in ³
$L = (0.5625 \cdot 0.3934 + 1) / 1.7651 + 0.5625^3 / 6.9022$	0.7177

Division 2 4.16.7.2 Step 5 - Flange Forces	
$H_D = 0.785 \cdot B^2 \cdot P$	
$H = 0.785 \cdot G^2 \cdot P$	
$H_T = H - H_D$	
$H_G = W_o - H$	
Results	
$H_D = 0.785 \cdot 12^2 \cdot 166.57$	18,829.16 lb _f
$H = 0.785 \cdot 12.5^2 \cdot 166.57$	20,430.95 lb _f
$H_T = 20,430.95 - 18,829.16$	1,601.79 lb _f
$H_G = 26,972.17 - 20,430.95$	6,541.22 lb _f

Division 2 4.16.7.2 Step 6 - Flange Moments of Inertia	
$I = 0.0874 \cdot L \cdot g_o^2 \cdot h_o \cdot B / V$	
$A_R = 0.5 \cdot (A - B)$	
$G_{avg} = 0.5 \cdot (g_o + g_i)$	
$t \geq G_{avg}$	
$A_A = A_R, \quad B_B = t, \quad C_C = h, \quad D_{DG} = G_{avg}$	
$K_{AB} = (A_A \cdot B_B^3) \cdot [1 / 3 - 0.21 \cdot (B_B / A_A) \cdot (1 - 1 / 12 \cdot \{B_B / A_A\}^4)]$	
$K_{CD} = (C_C \cdot D_{DG}^3) \cdot [1 / 3 - 0.105 \cdot (D_{DG} / C_C) \cdot (1 - 1 / 192 \cdot \{D_{DG} / C_C\}^4)]$	
$I_p = K_{AB} + K_{CD}$	
Results	
$I = 0.0874 \cdot 0.7177 \cdot 0.375^2 \cdot 2.1213 \cdot 12 / 0.2972$	0.7554 in ⁴
$A_R = 0.5 \cdot (16.5 - 12)$	2.25"
$G_{avg} = 0.5 \cdot (0.375 + 0.5625)$	0.4688"
$t = 0.5625 \text{ in} \geq G_{avg} = 0.4688 \text{ in}$	
$A_A = 2.25 \text{ in}, \quad B_B = 0.5625 \text{ in}, \quad C_C = 1.25 \text{ in}, \quad D_{DG} = 0.4688 \text{ in}$	
$K_{AB} = (2.25 \cdot 0.5625^3) \cdot [1 / 3 - 0.21 \cdot (0.5625 / 2.25) \cdot (1 - 1 / 12 \cdot \{0.5625 / 2.25\}^4)]$	0.1125 in ⁴
$K_{CD} = (1.25 \cdot 0.4688^3) \cdot [1 / 3 - 0.105 \cdot (0.4688 / 1.25) \cdot (1 - 1 / 192 \cdot \{0.4688 / 1.25\}^4)]$	0.0378 in ⁴
$I_p = 0.1125 + 0.0378$	0.1503 in ⁴

Division 2 4.16.7.2 Step 6 - Flange Moment Arms	
$h_D = (C - B - g_1) / 2$	
$h_T = 1 / 2 * [(C - B) / 2 + h_G]$	
$h_G = (C - G) / 2$	
Results	
$h_D = (15 - 12 - 0.5625) / 2$	1.2188"
$h_T = 1 / 2 * [(15 - 12) / 2 + 1.25]$	1.375"
$h_G = (15 - 12.5) / 2$	1.25"

Division 2 4.16.7.2 Step 6 - Flange Moment for Operating	
$M_{oe} = 4 * M_E * [I / (0.3846 * I_p + I)] * [h_D / (C - 2 * h_D)] + F_A * h_D$	(4.16.16)
$M_o = \text{abs} [(H_D * h_D + H_T * h_T + H_G * h_G + M_{oe}) * F_S]$	(4.16.14)
Results	
$M_{oe} = 4 * 0 * [0.7554 / (0.3846 * 0.1503 + 0.7554)] * [1.2188 / (15 - 2 * 1.2188)] + 0 * 1.2188$	0 lb _f -in
$M_o = \text{abs} [(18,829.16 * 1.2188 + 1,601.79 * 1.375 + 6,541.22 * 1.25 + 0) * 1]$	33,327 lb _f -in

Division 2 4.16.7.2 Step 7 - Flange Moment for Gasket Seating	
$M_g = W_g * (C - G) * F_S / 2$	(4.16.17)
Results	
$M_g = 33,686.09 * (15 - 12.5) * 1 / 2$	42,107.6 lb _f -in

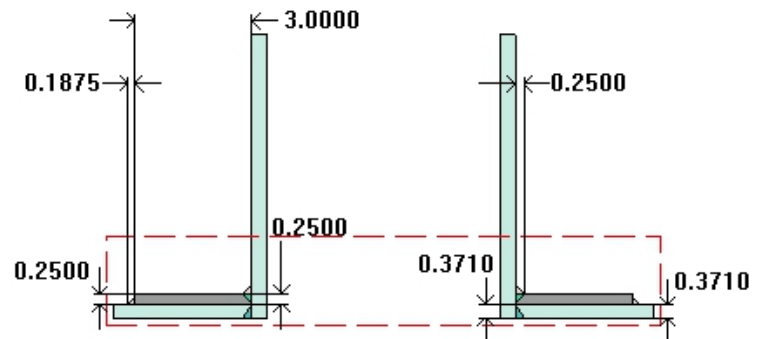
Division 2 4.16.7.2 Step 8 - Flange Stresses			
Operating		Gasket Seating	
$S_H = f \cdot M_o / (L \cdot g_1^{2 \cdot B})$		$S_H = f \cdot M_g / (L \cdot g_1^{2 \cdot B})$	
$S_R = (1.33 \cdot t \cdot e + 1) \cdot M_o / (L \cdot t^2 \cdot B)$		$S_R = (1.33 \cdot t \cdot e + 1) \cdot M_g / (L \cdot t^2 \cdot B)$	
$S_T = Y \cdot M_o / (t^2 \cdot B) - Z \cdot S_R$		$S_T = Y \cdot M_g / (t^2 \cdot B) - Z \cdot S_R$	
$S_H \leq \min[1.5 \cdot S_{fo}, 2.5 \cdot S_{no}]$		$S_H \leq \min[1.5 \cdot S_{fg}, 2.5 \cdot S_{ng}]$	
$S_H \leq 1.5 \cdot S_{fo}$		$S_H \leq 1.5 \cdot S_{fg}$	
$S_R \leq S_{fo}$		$S_R \leq S_{fg}$	
$S_T \leq S_{fo}$		$S_T \leq S_{fg}$	
$(S_H + S_R) / 2 \leq S_{fo}$		$(S_H + S_R) / 2 \leq S_{fg}$	
$(S_H + S_T) / 2 \leq S_{fo}$		$(S_H + S_T) / 2 \leq S_{fg}$	
$S_{fo} = 18,800 \text{ psi}$		$S_{fg} = 20,000 \text{ psi}$	
$S_{no} = 17,100 \text{ psi}$		$S_{ng} = 17,100 \text{ psi}$	
Operating			
	Stress (psi)	Allowable (psi)	Overstressed
$S_H = 1 \cdot 33,327 / (0.7177 \cdot 0.5625^2 \cdot 12)$	12,230	28,200	OK
$S_R = (1.33 \cdot 0.5625 \cdot 0.3934 + 1) \cdot 33,327 / (0.7177 \cdot 0.5625^2 \cdot 12)$	15,829	18,800	OK
$S_T = 6.2583 \cdot 33,327 / (0.5625^2 \cdot 12) - 3.2456 \cdot 15,829$	3,556	18,800	OK
$(S_H + S_R) / 2 = (12,230 + 15,829) / 2$	14,030	18,800	OK
$(S_H + S_T) / 2 = (12,230 + 3,556) / 2$	7,893	18,800	OK
Gasket Seating			
	Stress (psi)	Allowable (psi)	Overstressed
$S_H = 1 \cdot 42,107.6 / (0.7177 \cdot 0.5625^2 \cdot 12)$	15,452	30,000	OK
$S_R = (1.33 \cdot 0.5625 \cdot 0.3934 + 1) \cdot 42,107.6 / (0.7177 \cdot 0.5625^2 \cdot 12)$	20,000	20,000	OK
$S_T = 6.2583 \cdot 42,107.6 / (0.5625^2 \cdot 12) - 3.2456 \cdot 20,000$	4,493	20,000	OK
$(S_H + S_R) / 2 = (15,452 + 20,000) / 2$	17,726	20,000	OK
$(S_H + S_T) / 2 = (15,452 + 4,493) / 2$	9,973	20,000	OK

Division 2 4.16.7.2 Step 10 - Flange Rigidity			
Operating		Gasket Seating	
J = 52.14*V*M _o / (L*E _{yo} *g _o ² *K _R *h _o) ≤ 1	J = 52.14*V*M _g / (L*E _{yg} *g _o ² *K _R *h _o) ≤ 1		
K _R = 0.3			
Operating			
J = 52.14*0.2972*33,327 / (0.7177*26.0E+06*0.375 ² *0.3*2.1213)	0.3093	≤ 1	OK
Gasket Seating			
J = 52.14*0.2972*42,107.6 / (0.7177*29.4E+06*0.375 ² *0.3*2.1213)	0.3456	≤ 1	OK

Nozzle #1 (N1)

ASME Section VIII Division 1, 2013 Edition (Code Case 2695)					
Component	Nozzle				
Nozzle Material	SA-106 B Smls pipe (II-D p. 14, ln. 19)				
Pad Material	SA-516 70 (II-D p. 22, ln. 6)				
	Impact Tested	Normalized	Fine Grain Practice	PWHT	Optimize MDMT/ Find MAWP
Nozzle	No	No	No	No	No
Pad	No	No	No	No	No
	Design Pressure (psi)		Design Temperature (° F)		Design MDMT (° F)
Internal	100		650		-20
Location and Orientation					
Attached to	Cylinder #1				
Orientation	radial				
Offset, L	48"				
Angle, θ	90 °				
Distance, r	48"				
Through a Category A Joint	No				

Dimensions		
Description	NPS 12 Std Weight	
Inside Diameter	12"	
Nominal Wall Thickness	0.375"	
Local Vessel Minimum Thickness	0.371"	
Nozzle to Pad Groove Weld, t_{w2}	0.25"	
Leg ₄₁	0.25"	
Leg ₄₂	0.1875"	
Pad Width, W	3"	
Pad Thickness, t_e	0.25"	
External Projection Available, L_{pr1}	15.6875"	
Projection outside vessel to flange face, L_f	17.5"	
Corrosion	Inner	0"
	Outer	0"



Summary Tables

Division 2 4.5.5 Maximum local primary membrane stress					
Condition	P (psi)	P _L (psi)	S _{allow} (psi)	Over stressed	P _{max} (psi)
Design P	100	<u>16.033</u>	<u>28.200</u>	No	<u>175.89</u>

Division 2 4.5.4.1 Nozzle Neck Minimum Thickness								
Condition	P (psi)	t _{d1} (in)	t _{d,STD} (in)	t (in)	t _d (in)	t _n (in)	Status	P _{max} (psi)
Design P	100	<u>0.0352</u>	0.3281	0.371	0.3281	0.375	OK	<u>1.036.68</u>

Division 2 4.5.14 Strength of Nozzle Attachment Welds Summary											
Condition	P (psi)	k _y	L _τ (in)	L _{τp} (in)	f _{welds} (lb _f)	τ ₁ (psi)	τ ₂ (psi)	τ ₃ (psi)	τ (psi)	S (psi)	Over stressed
Design P	100	1.0625	10.0138	14.7262	<u>3.191.91</u>	<u>855</u>	<u>542</u>	<u>1.343</u>	<u>1.343</u>	18,800	No
MAWP	175.89	1.0625	10.0138	14.7262	<u>5.614.26</u>	<u>1.505</u>	<u>954</u>	<u>2.363</u>	<u>2.363</u>	18,800	No

Division 2 Table 4.2.11 Minimum Fillet Weld Sizes			
Fillet Leg	Required Leg Size (in)	Actual Leg Size (in)	Status
Leg ₄₁	0.3571	0.25	weld size is NOT adequate
Leg ₄₂	0.2143	0.1875	weld size is NOT adequate

WRC 107												
Load Case	P (psi)	P _r (lb _f)	M _c (lb _f -in)	V _c (lb _f)	M _L (lb _f -in)	V _L (lb _f)	M _t (lb _f -in)	Max Comb Stress (psi)	Allow Comb Stress (psi)	Max Local Primary Stress (psi)	Allow Local Primary Stress (psi)	Over stressed
Operating	100	600	500	400	300	200	100	16,148	56,400	15,827	28,200	No
Operating (Hot Shut Down)	0	600	500	400	300	200	100	-2,050	56,400	-452	28,200	No
Start up	100	100	200	300	400	500	600	16,060	56,400	16,015	28,200	No
Start up (Hot Shut Down)	0	100	200	300	400	500	600	-392	56,400	-90	28,200	No

WRC 107 MAWP												
Load Case	P (psi)	P _r (lbf _f)	M _c (lbf-in)	V _c (lbf _f)	M _L (lbf-in)	V _L (lbf _f)	M _t (lbf-in)	Max Comb Stress (psi)	Allow Comb Stress (psi)	Max Local Primary Stress (psi)	Allow Local Primary Stress (psi)	Over stressed
Operating	176.01	600	500	400	300	200	100	28,333	56,400	28,012	28,200	No
Operating (Hot Shut Down)	0	600	500	400	300	200	100	-2,050	56,400	-452	28,200	No
Start up	176.01	100	200	300	400	500	600	28,245	56,400	28,200	28,200	No
Start up (Hot Shut Down)	0	100	200	300	400	500	600	-392	56,400	-90	28,200	No

Detailed Calculations

Division 2 4.5.5 Vessel effective radius	
$R_{eff} = 0.5 \cdot D_i$	(4.5.3)
$R_{xn} = t_n / \ln[1 + t_n / R_n]$	(4.5.51)
Corroded\New	
$R_{eff} = 0.5 \cdot 60.258 =$	30.129"
$R_{xn} = 0.375 / \ln[1 + 0.375 / 6] =$	6.1856"

Division 2 4.5.5 Limit of reinforcement along the vessel wall	
$L_{R1} = (R_{eff} \cdot t)^{0.5} + W$	(4.5.6)
$L_{R2} = ((R_{eff} + t) \cdot (t + t_e))^{0.5}$	(4.5.7)
$L_{R3} = 2 \cdot R_n$	(4.5.8)
$L_R = \min[L_{R1}, L_{R2}, L_{R3}]$	(4.5.9)
Corroded\New	
$L_{R1} = (30.129 \cdot 0.371)^{0.5} + 3 =$	6.3433"
$L_{R2} = ((30.129 + 0.371) \cdot (0.371 + 0.25))^{0.5} =$	4.3521"
$L_{R3} = 2 \cdot 6 =$	12"
$L_R = \min[6.3433, 4.3521, 12] =$	4.3521"

Division 2 4.5.5 Limit of reinforcement along the nozzle wall projecting outside the vessel surface	
$L_{H1} = \min[1.5 \cdot t, t_e] + (R_n \cdot t_n)^{0.5}$	(4.5.11)
$L_{H2} = L_{pr1}$	(4.5.12)
$L_{H3} = 8 \cdot (t + t_e)$	(4.5.13)
$L_H = \min[L_{H1}, L_{H2}, L_{H3}] + t$	(4.5.14)
Corroded\New	
$L_{H1} = \min[1.5 \cdot 0.371, 0.25] + (6 \cdot 0.375)^{0.5} =$	1.75"
$L_{H2} = 15.6875 =$	15.6875"
$L_{H3} = 8 \cdot (0.371 + 0.25) =$	4.968"
$L_H = \min[1.75, 15.6875, 4.968] + 0.371 =$	2.121"

Division 2 4.5.5 Available area near the nozzle opening	
$\lambda = \min[(2 \cdot R_n + t_n) / ((D_i + t_{eff}) \cdot t_{eff})^{0.5}, 12]$	(4.5.24)
$A_1 = t \cdot L_R \cdot \max[(\lambda / 5)^{0.85}, 1]$	(4.5.23)
$A_2 = t_n \cdot L_H$	(4.5.27)
$A_{41} = 0.5 \cdot L_{41}^2$	(4.5.39)
$A_{42} = 0.5 \cdot L_{42}^2$	(4.5.40)
$A_{5a} = W \cdot t_e$	(4.5.42)
$A_{5b} = L_R \cdot t_e$	(4.5.43)
$A_5 = \min[A_{5a}, A_{5b}]$	(4.5.45)
Corroded\New	
$\lambda = \min[(2 \cdot 6 + 0.375) / ((60.258 + 0.5433) \cdot 0.5433)^{0.5}, 12] =$	2.1531
$A_1 = 0.371 \cdot 4.3521 \cdot \max[(2.1531 / 5)^{0.85}, 1] =$	1.6146 in ²
$A_2 = 0.375 \cdot 2.121 =$	0.7954 in ²
$A_{41} = 0.5 \cdot 0.25^2 =$	0.0313 in ²
$A_{42} = 0.5 \cdot 0.1875^2 =$	0.0176 in ²
$A_{5a} = 3 \cdot 0.25 =$	0.75 in ²
$A_{5b} = 4.3521 \cdot 0.25 =$	1.088 in ²
$A_5 = \min[0.75, 1.088] =$	0.75 in ²

Division 2 4.5.5 Equations	
$f_m = \min[S_n / S , 1]$	(4.5.21)
$f_{rp} = \min[S_p / S , 1]$	(4.5.22)
$t_{eff} = t + (A_5 * f_{rp} / L_R)$	(4.5.25)
$R_{xs} = t_{eff} / \ln[1 + t_{eff} / R_{eff}]$	(4.5.52)
$A_T = A_1 + f_m * A_2 + A_{41} + A_{42} + f_{rp} * A_5$	(4.5.20)
$f_N = P * R_{xn} * L_H$	(4.5.46)
$f_S = P * R_{xs} * (L_R + t_n)$	(4.5.48)
$f_Y = P * R_{xs} * R_{nc}$	(4.5.50)
$\sigma_{avg} = (f_N + f_S + f_Y) / A_T$	(4.5.53)
$\sigma_{circ} = P * R_{xs} / t_{eff}$	(4.5.54)
$P_L = \max[\{2 * \sigma_{avg} - \sigma_{circ}\} , \sigma_{circ}]$	(4.5.55)
$S_{allow} = 1.5 * S * E$	(4.5.57)
$P_L \leq S_{allow}$	(4.5.56)
$A_p = (f_N + f_S + f_Y) / P$	(4.5.62)
$P_{max1} = S_{allow} / (2 * A_p / A_T - R_{xs} / t_{eff})$	(4.5.59)
$P_{max2} = S * (t / R_{xs})$	(4.5.60)
$P_{max} = \min[P_{max1} , P_{max2}]$	(4.5.61)

Division 2 4.5.5 Design Rules for Openings in Shells and Heads	
Design P	
Total available area near the nozzle opening	
$f_m = \min[17,100 / 18,800 , 1] =$	0.9096
$f_{rp} = \min[18,800 / 18,800 , 1] =$	1
$t_{eff} = 0.371 + (0.75 \cdot 1 / 4.3521) =$	0.5433"
$R_{xs} = 0.5433 / \ln[1 + 0.5433 / 30.129] =$	30.3999"
$A_T = 1.6146 + 0.9096 \cdot 0.7954 + 0.0313 + 0.0176 + 1 \cdot 0.75 =$	3.1369 in ²
Forces at nozzle to vessel intersection	
$f_N = 100 \cdot 6.1856 \cdot 2.121 =$	1,311.97 lb _f
$f_S = 100 \cdot 30.3999 \cdot (4.3521 + 0.375) =$	14,370.22 lb _f
$f_Y = 100 \cdot 30.3999 \cdot 6 =$	18,239.91 lb _f
Average local primary membrane stress	
$\sigma_{avg} = (1,311.97 + 14,370.22 + 18,239.91) / 3.1369 =$	10,814 psi
General primary membrane stress	
$\sigma_{circ} = 100 \cdot 30.3999 / 0.5433 =$	5,595 psi
Maximum local primary membrane stress at the nozzle intersection	
$P_L = \max[\{2 \cdot 10,814 - 5,595\} , 5,595] =$	16.033 psi
Allowable stress	
$S_{allow} = 1.5 \cdot 18,800 \cdot 1 =$	28.200 psi
$P_L = 16,033 \text{ psi} \leq S_{allow} = 28,200 \text{ psi}$	
Maximum allowable working pressure	
$A_p = (1,311.97 + 14,370.22 + 18,239.91) / 100 =$	339.221 in ²
$P_{max1} = 28,200 / (2 \cdot 339.221 / 3.1369 - 30.3999 / 0.5433) =$	175.89 psi
$P_{max2} = 18,800 \cdot (0.371 / 30.3999) =$	229.44 psi
$P_{max} = \min[175.89, 229.44] =$	175.89 psi

Division 2 4.5.14 Equations	
$k_y = (R_{nc} + t_n) / R_{nc}$	(4.5.174)
$L_\tau = \pi / 2 * (R_n + t_n)$	(4.5.175)
$L_{\tau p} = \pi / 2 * (R_n + t_n + W)$	(4.5.177)
$L_{41T} = 0.7071 * L_{41}$	(4.5.179)
$L_{42T} = 0.7071 * L_{42}$	(4.5.180)
$f_{welds} = \min[f_y * k_y, 1.5 * S_n * (A_2 + A_3), \pi / 4 * P * R_n^2 * k_y^2]$	(4.5.184)
$f_{ws} = f_{welds} * t * S / (t * S + t_e * S_p)$	(4.5.189)
$f_{wp} = f_{welds} * t_e * S_p / (t * S + t_e * S_p)$	(4.5.190)
$\tau_1 = f_{ws} / [L_\tau * (0.6 * t_{w1} + 0.49 * L_{43T})]$	(4.5.185)
$\tau_2 = f_{wp} / [L_\tau * (0.6 * t_{w2} + 0.49 * L_{41T})]$	(4.5.186)
$\tau_3 = f_{wp} / [L_{\tau p} * 0.49 * L_{42T}]$	(4.5.187)
$\tau = \max[\tau_1, \tau_2, \tau_3]$	
$\max[\tau_1, \tau_2, \tau_3] \leq S$	(4.5.188)

Division 2 4.5.14 Strength of Nozzle Attachment Welds	
Corroded\New	
Discontinuity force factor	
$k_y = (6 + 0.375) / 6 =$	1.0625
Weld length resisting discontinuity force	
$L_\tau = \pi / 2 * (6 + 0.375) =$	10.0138"
$L_{\tau p} = \pi / 2 * (6 + 0.375 + 3) =$	14.7262"
Weld throat dimensions	
$L_{41T} = 0.7071 * 0.25 =$	0.1768"
$L_{42T} = 0.7071 * 0.1875 =$	0.1326"
Design P	
Average shear stress in weld	
$f_{welds} = \min[18,239.91 * 1.0625, 1.5 * 17,100 * (0.7954 + 0) , \pi / 4 * 100 * 6^2 * 1.0625^2] =$	3,191.91 lb_f
$f_{ws} = 3,191.91 * 0.371 * 18,800 / (0.371 * 18,800 + 0.25 * 18,800) =$	1,906.92 lb _f
$f_{wp} = 3,191.91 * 0.25 * 18,800 / (0.371 * 18,800 + 0.25 * 18,800) =$	1,284.99 lb _f
$\tau_1 = 1,906.92 / [10.0138 * (0.6 * 0.371 + 0.49 * 0)] =$	855 psi
$\tau_2 = 1,284.99 / [10.0138 * (0.6 * 0.25 + 0.49 * 0.1768)] =$	542 psi
$\tau_3 = 1,284.99 / [14.7262 * 0.49 * 0.1326] =$	1,343 psi
$\tau = \max[855, 542, 1,343] =$	1,343 psi
$\tau = 1,343 \text{ psi} \leq S = 18,800 \text{ psi}$	
MAWP	
Average shear stress in weld	
$f_{welds} = \min[32,082.25 * 1.0625, 1.5 * 17,100 * (0.7954 + 0) , \pi / 4 * 175.89 * 6^2 * 1.0625^2] =$	5,614.26 lb_f
$f_{ws} = 5,614.26 * 0.371 * 18,800 / (0.371 * 18,800 + 0.25 * 18,800) =$	3,354.09 lb _f
$f_{wp} = 5,614.26 * 0.25 * 18,800 / (0.371 * 18,800 + 0.25 * 18,800) =$	2,260.17 lb _f
$\tau_1 = 3,354.09 / [10.0138 * (0.6 * 0.371 + 0.49 * 0)] =$	1,505 psi
$\tau_2 = 2,260.17 / [10.0138 * (0.6 * 0.25 + 0.49 * 0.1768)] =$	954 psi
$\tau_3 = 2,260.17 / [14.7262 * 0.49 * 0.1326] =$	2,363 psi
$\tau = \max[1,505, 954, 2,363] =$	2,363 psi
$\tau = 2,363 \text{ psi} \leq S = 18,800 \text{ psi}$	

Division 2 4.3.3.1 Design Thickness for Internal Pressure	
$t_d = D / 2 * (\exp[P / (S * E)] - 1) + \text{Corrosion}$	(4.3.1)
Operating Hot & Corroded	
$t_d = 12 / 2 * (\exp[(100 + 0) / (17,100 * 1)] - 1) + 0 =$	0.0352"
Operating Cold & Corroded	
$t_r = 12 / 2 * (\exp[(144.02 + 0) / (17,100 * 1)] - 1) =$	0.0507"

Division 2 4.3.3.1 Maximum Pressure Circumferential Stress	
$P_{\max} = S * E * \ln[D_o / D]$	(4.3.1)
MAWP	
$P_{\max} = 17,100 * 1 * \ln[12.75 / 12] - 0 =$	1,036.68 psi

Division 2 4.3.10.2 Combined Loads Summary (Operating governs)									
				$\sigma_{\theta m}$ (psi)	σ_{sm} (psi)	σ_T (psi)	S_{allow} (psi)	t_d (in)	$P_{\max}$ (psi)
Operating Hot & Corroded	Weight Only	Left	Windward	1,600	748	1,429	17,100	0.0322	1,196.69
			Leeward	1,600	721	1,430	17,100	0.0322	1,196.69

WRC 107 Operating

Applied Loads

Radial load:	$P_r = 600$	lb_f
Circumferential moment:	$M_c = 500$	$\text{lb}_f\text{-in}$
Circumferential shear:	$V_c = 400$	lb_f
Longitudinal moment:	$M_L = 300$	$\text{lb}_f\text{-in}$
Longitudinal shear:	$V_L = 200$	lb_f
Torsion moment:	$M_t = 100$	$\text{lb}_f\text{-in}$
Internal pressure:	$P = 176.006$	psi
Mean shell radius:	$R_m = 30.3145$	in
Local shell thickness:	$T = 0.371$	in
Shell yield stress:	$S_y = 28,200$	psi
Design factor:	3	

Maximum stresses due to the applied loads at the pad edge (includes pressure)

$$\gamma = R_m / T = 30.3145 / 0.371 = 81.7102$$

$$\beta = 0.875 * r_o / R_m = 0.875 * 9.375 / 30.3145 = 0.2706$$

Pressure stress intensity factor, $I = 1.0883$ (derived from PVP-Vol. 399, pages 77-82)

$$\text{Local circumferential pressure stress} = I * P * R_i / T = 15,555 \text{ psi}$$

$$\text{Local longitudinal pressure stress} = I * P * R_i / (2 * T) = 7,777 \text{ psi}$$

$$\text{Maximum combined stress } (P_L + P_b + Q) = 15,989 \text{ psi}$$

$$\text{Allowable combined stress } (P_L + P_b + Q) = \pm 3 * S = \pm 56,400 \text{ psi}$$

The maximum combined stress $(P_L + P_b + Q)$ is within allowable limits.

$$\text{Maximum local primary membrane stress } (P_L) = 15,160 \text{ psi}$$

$$\text{Allowable local primary membrane stress } (P_L) = \pm 1.5 * S = \pm 28,200 \text{ psi}$$

The maximum local primary membrane stress (P_L) is within allowable limits.

Stresses at the pad edge per WRC Bulletin 107										
Figure	value		A _u	A _l	B _u	B _l	C _u	C _l	D _u	D _l
3C*	N _φ / (P / R _m)	3.0951	0	0	0	0	-165	-165	-165	-165
4C*	N _φ / (P / R _m)	7.7129	-411	-411	-411	-411	0	0	0	0
1C	M _φ / P	0.0659	0	0	0	0	-1,724	1,724	-1,724	1,724
2C-1	M _φ / P	0.012	-314	314	-314	314	0	0	0	0
3A*	N _φ / [M _C / (R _m ² *β)]	2.3025	0	0	0	0	-12	-12	12	12
1A	M _φ / [M _C / (R _m ² *β)]	0.0559	0	0	0	0	-149	149	149	-149
3B*	N _φ / [M _L / (R _m ² *β)]	4.8813	-16	-16	16	16	0	0	0	0
1B-1	M _φ / [M _L / (R _m ² *β)]	0.0109	-17	17	17	-17	0	0	0	0
Pressure stress*			15,555	15,555	15,555	15,555	14,293	14,293	14,293	14,293
Total circumferential stress			14,797	15,459	14,863	15,457	12,243	15,989	12,565	15,715
Primary membrane circumferential stress*			15,128	15,128	15,160	15,160	14,116	14,116	14,140	14,140
3C*	N _x / (P / R _m)	3.0951	-165	-165	-165	-165	0	0	0	0
4C*	N _x / (P / R _m)	7.7129	0	0	0	0	-411	-411	-411	-411
1C-1	M _x / P	0.0294	-769	769	-769	769	0	0	0	0
2C	M _x / P	0.0312	0	0	0	0	-816	816	-816	816
4A*	N _x / [M _C / (R _m ² *β)]	7.5318	0	0	0	0	-41	-41	41	41
2A	M _x / [M _C / (R _m ² *β)]	0.0232	0	0	0	0	-62	62	62	-62
4B*	N _x / [M _L / (R _m ² *β)]	2.5238	-8	-8	8	8	0	0	0	0
2B-1	M _x / [M _L / (R _m ² *β)]	0.0182	-29	29	29	-29	0	0	0	0
Pressure stress*			7,146	7,146	7,146	7,146	7,777	7,777	7,777	7,777
Total longitudinal stress			6,175	7,771	6,249	7,729	6,447	8,203	6,653	8,161
Primary membrane longitudinal stress*			6,973	6,973	6,989	6,989	7,325	7,325	7,407	7,407
Shear from M _t			0	0	0	0	0	0	0	0
Circ shear from V _C			37	37	-37	-37	0	0	0	0
Long shear from V _L			0	0	0	0	-18	-18	18	18
Total Shear stress			37	37	-37	-37	-18	-18	18	18
Combined stress (P _L +P _b +Q)			14,797	15,459	14,863	15,457	12,243	15,989	12,565	15,715

Note: * denotes primary stress.

Maximum stresses due to the applied loads at the nozzle OD (includes pressure)

$$\gamma = R_m / T = 30.3145 / 0.621 = 48.8156$$

$$\beta = 0.875 * r_o / R_m = 0.875 * 6.375 / 30.3145 = 0.184$$

Pressure stress intensity factor, I = 1.9742 (derived from PVP-Vol. 399, pages 77-82)

$$\text{Local circumferential pressure stress} = I * P * R_i / T = 28,218 \text{ psi}$$

$$\text{Local longitudinal pressure stress} = I * P * R_i / (2 * T) = 14,108 \text{ psi}$$

$$\text{Maximum combined stress } (P_L + P_b + Q) = 28,333 \text{ psi}$$

$$\text{Allowable combined stress } (P_L + P_b + Q) = +3 * S = +56,400 \text{ psi}$$

The maximum combined stress (P_L+P_b+Q) is within allowable limits.

Maximum local primary membrane stress (P_L) = 28,012 psi

Allowable local primary membrane stress (P_L) = $\pm 1.5S = \pm 28,200$ psi

The maximum local primary membrane stress (P_L) is within allowable limits.

Stresses at the nozzle OD per WRC Bulletin 107										
Figure	value		A _u	A _l	B _u	B _l	C _u	C _l	D _u	D _l
3C*	N _φ / (P / R _m)	4.6113	0	0	0	0	-147	-147	-147	-147
4C*	N _φ / (P / R _m)	6.935	-221	-221	-221	-221	0	0	0	0
1C	M _φ / P	0.0688	0	0	0	0	-642	642	-642	642
2C-1	M _φ / P	0.0354	-330	330	-330	330	0	0	0	0
3A*	N _φ / [M _c / (R _m ² *β)]	2.0159	0	0	0	0	-10	-10	10	10
1A	M _φ / [M _c / (R _m ² *β)]	0.0749	0	0	0	0	-104	104	104	-104
3B*	N _φ / [M _L / (R _m ² *β)]	5.0821	-15	-15	15	15	0	0	0	0
1B-1	M _φ / [M _L / (R _m ² *β)]	0.0254	-21	21	21	-21	0	0	0	0
Pressure stress*			28,218	28,218	28,218	28,218	14,293	14,293	14,293	14,293
Total circumferential stress			27,631	28,333	27,703	28,321	13,390	14,882	13,618	14,694
Primary membrane circumferential stress*			27,982	27,982	28,012	28,012	14,136	14,136	14,156	14,156
3C*	N _x / (P / R _m)	4.6113	-147	-147	-147	-147	0	0	0	0
4C*	N _x / (P / R _m)	6.935	0	0	0	0	-221	-221	-221	-221
1C-1	M _x / P	0.0681	-636	636	-636	636	0	0	0	0
2C	M _x / P	0.039	0	0	0	0	-364	364	-364	364
4A*	N _x / [M _c / (R _m ² *β)]	3.8603	0	0	0	0	-18	-18	18	18
2A	M _x / [M _c / (R _m ² *β)]	0.0371	0	0	0	0	-52	52	52	-52
4B*	N _x / [M _L / (R _m ² *β)]	2.0143	-6	-6	6	6	0	0	0	0
2B-1	M _x / [M _L / (R _m ² *β)]	0.0378	-32	32	32	-32	0	0	0	0
Pressure stress*			7,146	7,146	7,146	7,146	14,108	14,108	14,108	14,108
Total longitudinal stress			6,325	7,661	6,401	7,609	13,453	14,285	13,593	14,217
Primary membrane longitudinal stress*			6,993	6,993	7,005	7,005	13,869	13,869	13,905	13,905
Shear from M _t			1	1	1	1	1	1	1	1
Circ shear from V _c			32	32	-32	-32	0	0	0	0
Long shear from V _L			0	0	0	0	-16	-16	16	16
Total Shear stress			33	33	-31	-31	-15	-15	17	17
Combined stress (P _L +P _b +Q)			27,631	28,333	27,703	28,321	13,456	14,882	13,627	14,695

Note: * denotes primary stress.

Shear stress in the nozzle wall due to external loads

$$\begin{aligned}\sigma_{\text{shear}} &= (V_L^2 + V_C^2)^{0.5} / (\pi * R_i * t_n) \\ &= (200^2 + 400^2)^{0.5} / (\pi * 6 * 0.375) \\ &= 63 \text{ psi}\end{aligned}$$

$$\begin{aligned}\sigma_{\text{torsion}} &= M_t / (2 * \pi * R_i^2 * t_n) \\ &= 100 / (2 * \pi * 6^2 * 0.375) \\ &= 1 \text{ psi}\end{aligned}$$

$$\begin{aligned}\sigma_{\text{total}} &= \sigma_{\text{shear}} + \sigma_{\text{torsion}} \\ &= 63 + 1 \\ &= 64 \text{ psi}\end{aligned}$$

UG-45: The total combined shear stress (64 psi) ≤ allowable (0.7*S_n = 0.7*17,100 = 11,970 psi)

WRC 107 Start up

Applied Loads

Radial load: $P_r = 100 \text{ lb}_f$
Circumferential moment: $M_c = 200 \text{ lb}_f\text{-in}$
Circumferential shear: $V_c = 300 \text{ lb}_f$
Longitudinal moment: $M_L = 400 \text{ lb}_f\text{-in}$
Longitudinal shear: $V_L = 500 \text{ lb}_f$
Torsion moment: $M_t = 600 \text{ lb}_f\text{-in}$
Internal pressure: $P = 176.006 \text{ psi}$
Mean shell radius: $R_m = 30.3145 \text{ in}$
Local shell thickness: $T = 0.371 \text{ in}$
Shell yield stress: $S_y = 28,200 \text{ psi}$
Design factor: 3

Maximum stresses due to the applied loads at the pad edge (includes pressure)

$$\gamma = R_m / T = 30.3145 / 0.371 = 81.7102$$

$$\beta = 0.875 * r_o / R_m = 0.875 * 9.375 / 30.3145 = 0.2706$$

Pressure stress intensity factor, $I = 1.0883$ (derived from PVP-Vol. 399, pages 77-82)

$$\text{Local circumferential pressure stress} = I * P * R_i / T = 15,555 \text{ psi}$$

$$\text{Local longitudinal pressure stress} = I * P * R_i / (2 * T) = 7,777 \text{ psi}$$

$$\text{Maximum combined stress } (P_L + P_b + Q) = 15,540 \text{ psi}$$

$$\text{Allowable combined stress } (P_L + P_b + Q) = \pm 3 * S = \pm 56,400 \text{ psi}$$

The maximum combined stress $(P_L + P_b + Q)$ is within allowable limits.

$$\text{Maximum local primary membrane stress } (P_L) = 15,507 \text{ psi}$$

$$\text{Allowable local primary membrane stress } (P_L) = \pm 1.5 * S = \pm 28,200 \text{ psi}$$

The maximum local primary membrane stress (P_L) is within allowable limits.

Stresses at the pad edge per WRC Bulletin 107										
Figure	value		A _u	A _l	B _u	B _l	C _u	C _l	D _u	D _l
3C*	N _φ / (P / R _m)	3.0951	0	0	0	0	-28	-28	-28	-28
4C*	N _φ / (P / R _m)	7.7129	-69	-69	-69	-69	0	0	0	0
1C	M _φ / P	0.0659	0	0	0	0	-287	287	-287	287
2C-1	M _φ / P	0.012	-52	52	-52	52	0	0	0	0
3A*	N _φ / [M _c / (R _m ² *β)]	2.3025	0	0	0	0	-5	-5	5	5
1A	M _φ / [M _c / (R _m ² *β)]	0.0559	0	0	0	0	-59	59	59	-59
3B*	N _φ / [M _L / (R _m ² *β)]	4.8813	-21	-21	21	21	0	0	0	0
1B-1	M _φ / [M _L / (R _m ² *β)]	0.0109	-23	23	23	-23	0	0	0	0
Pressure stress*			15,555	15,555	15,555	15,555	14,293	14,293	14,293	14,293
Total circumferential stress			15,390	15,540	15,478	15,536	13,914	14,606	14,042	14,498
Primary membrane circumferential stress*			15,465	15,465	15,507	15,507	14,260	14,260	14,270	14,270
3C*	N _x / (P / R _m)	3.0951	-28	-28	-28	-28	0	0	0	0
4C*	N _x / (P / R _m)	7.7129	0	0	0	0	-69	-69	-69	-69
1C-1	M _x / P	0.0294	-128	128	-128	128	0	0	0	0
2C	M _x / P	0.0312	0	0	0	0	-136	136	-136	136
4A*	N _x / [M _c / (R _m ² *β)]	7.5318	0	0	0	0	-16	-16	16	16
2A	M _x / [M _c / (R _m ² *β)]	0.0232	0	0	0	0	-25	25	25	-25
4B*	N _x / [M _L / (R _m ² *β)]	2.5238	-11	-11	11	11	0	0	0	0
2B-1	M _x / [M _L / (R _m ² *β)]	0.0182	-39	39	39	-39	0	0	0	0
Pressure stress*			7,146	7,146	7,146	7,146	7,777	7,777	7,777	7,777
Total longitudinal stress			6,940	7,274	7,040	7,218	7,531	7,853	7,613	7,835
Primary membrane longitudinal stress*			7,107	7,107	7,129	7,129	7,692	7,692	7,724	7,724
Shear from M _t			3	3	3	3	3	3	3	3
Circ shear from V _c			27	27	-27	-27	0	0	0	0
Long shear from V _L			0	0	0	0	-46	-46	46	46
Total Shear stress			30	30	-24	-24	-43	-43	49	49
Combined stress (P _L +P _b +Q)			15,390	15,540	15,478	15,536	13,914	14,606	14,042	14,498

Note: * denotes primary stress.

Maximum stresses due to the applied loads at the nozzle OD (includes pressure)

$$\gamma = R_m / T = 30.3145 / 0.621 = 48.8156$$

$$\beta = 0.875 * r_o / R_m = 0.875 * 6.375 / 30.3145 = 0.184$$

Pressure stress intensity factor, I = 1.9742 (derived from PVP-Vol. 399, pages 77-82)

$$\text{Local circumferential pressure stress} = I * P * R_i / T = 28,218 \text{ psi}$$

$$\text{Local longitudinal pressure stress} = I * P * R_i / (2 * T) = 14,108 \text{ psi}$$

$$\text{Maximum combined stress } (P_L + P_b + Q) = 28,245 \text{ psi}$$

$$\text{Allowable combined stress } (P_L + P_b + Q) = +3 * S = +56,400 \text{ psi}$$

The maximum combined stress (P_L+P_b+Q) is within allowable limits.

Maximum local primary membrane stress (P_L) = 28,200 psi

Allowable local primary membrane stress (P_L) = $\pm 1.5S = \pm 28,200$ psi

The maximum local primary membrane stress (P_L) is within allowable limits.

Stresses at the nozzle OD per WRC Bulletin 107										
Figure	value		A _u	A _l	B _u	B _l	C _u	C _l	D _u	D _l
3C*	N _φ / (P / R _m)	4.6113	0	0	0	0	-24	-24	-24	-24
4C*	N _φ / (P / R _m)	6.935	-37	-37	-37	-37	0	0	0	0
1C	M _φ / P	0.0688	0	0	0	0	-107	107	-107	107
2C-1	M _φ / P	0.0354	-55	55	-55	55	0	0	0	0
3A*	N _φ / [M _c / (R _m ² *β)]	2.0159	0	0	0	0	-4	-4	4	4
1A	M _φ / [M _c / (R _m ² *β)]	0.0749	0	0	0	0	-42	42	42	-42
3B*	N _φ / [M _L / (R _m ² *β)]	5.0821	-19	-19	19	19	0	0	0	0
1B-1	M _φ / [M _L / (R _m ² *β)]	0.0254	-28	28	28	-28	0	0	0	0
Pressure stress*			28,218	28,218	28,218	28,218	14,293	14,293	14,293	14,293
Total circumferential stress			28,079	28,245	28,173	28,227	14,116	14,414	14,208	14,338
Primary membrane circumferential stress*			28,162	28,162	28,200	28,200	14,265	14,265	14,273	14,273
3C*	N _x / (P / R _m)	4.6113	-24	-24	-24	-24	0	0	0	0
4C*	N _x / (P / R _m)	6.935	0	0	0	0	-37	-37	-37	-37
1C-1	M _x / P	0.0681	-106	106	-106	106	0	0	0	0
2C	M _x / P	0.039	0	0	0	0	-61	61	-61	61
4A*	N _x / [M _c / (R _m ² *β)]	3.8603	0	0	0	0	-7	-7	7	7
2A	M _x / [M _c / (R _m ² *β)]	0.0371	0	0	0	0	-21	21	21	-21
4B*	N _x / [M _L / (R _m ² *β)]	2.0143	-8	-8	8	8	0	0	0	0
2B-1	M _x / [M _L / (R _m ² *β)]	0.0378	-42	42	42	-42	0	0	0	0
Pressure stress*			7,146	7,146	7,146	7,146	14,108	14,108	14,108	14,108
Total longitudinal stress			6,966	7,262	7,066	7,194	13,982	14,146	14,038	14,118
Primary membrane longitudinal stress*			7,114	7,114	7,130	7,130	14,064	14,064	14,078	14,078
Shear from M _t			4	4	4	4	4	4	4	4
Circ shear from V _c			24	24	-24	-24	0	0	0	0
Long shear from V _L			0	0	0	0	-40	-40	40	40
Total Shear stress			28	28	-20	-20	-36	-36	44	44
Combined stress (P _L +P _b +Q)			28,079	28,245	28,173	28,227	14,125	14,419	14,219	14,346

Note: * denotes primary stress.

Shear stress in the nozzle wall due to external loads

$$\begin{aligned}\sigma_{\text{shear}} &= (V_L^2 + V_C^2)^{0.5} / (\pi^* R_i^* t_n) \\ &= (500^2 + 300^2)^{0.5} / (\pi^* 6^* 0.375) \\ &= 82 \text{ psi}\end{aligned}$$

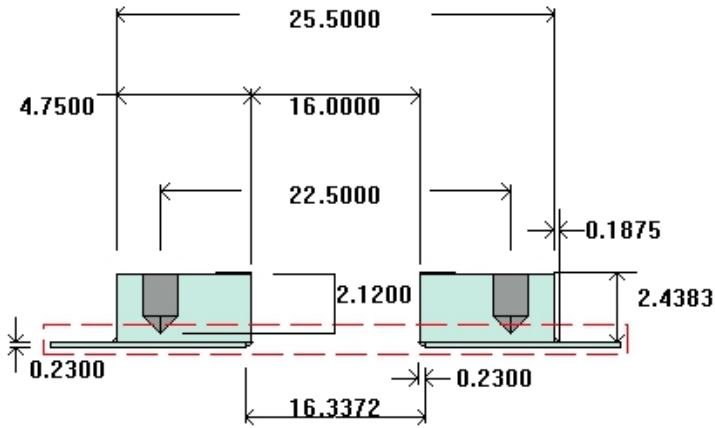
$$\begin{aligned}\sigma_{\text{torsion}} &= M_t / (2^* \pi^* R_i^2 * t_n) \\ &= 600 / (2^* \pi^* 6^2 * 0.375) \\ &= 7 \text{ psi}\end{aligned}$$

$$\begin{aligned}\sigma_{\text{total}} &= \sigma_{\text{shear}} + \sigma_{\text{torsion}} \\ &= 82 + 7 \\ &= 90 \text{ psi}\end{aligned}$$

UG-45: The total combined shear stress (90 psi) ≤ allowable (0.7*S_n = 0.7*17,100 = 11,970 psi)

Nozzle #2 (N2)

ASME Section VIII Division 1, 2013 Edition



Pad inner diameter =	16 in
Shell opening diameter =	16.3372 in
Pad thickness =	2.5 in
Tapped hole diameter =	1.25 in
Tapped hole depth =	2.12 in
Tapped hole bolt circle =	22.5 in
Raised face height =	0.06 in
Raised face outer diameter =	18.5 in
Inner fillet =	0.1875 in
Lower fillet =	0.23 in
D_p =	25.5 in
t_e =	2.44 in

Note: round inside edges per UG-76(c)

Note: Thread engagement shall comply with the requirements of UG-43(g).

Location and Orientation

Located on:	Ellipsoidal Head #1
Orientation:	0°
End of nozzle to datum line:	210.2474 in
Calculated as hillside:	No
Distance to head center, R:	0 in
Passes through a Category A joint:	No

Nozzle

Access opening:	No
Material specification:	SA-105 (II-D p. 18, In. 32)
Bolt material specification:	SA-193 B7 Bolt <= 2 1/2 (II-D p. 352, In. 31)
Description:	Studding Outlet NPS 16 Class 300
Corrosion allowance:	0 in
Projection available outside vessel, Lpr:	2.3783 in
User input vessel thickness:	0.23 in
Liquid static head included:	0 psi
Longitudinal joint efficiency:	1

Reinforcement Calculations for MAWP

The vessel wall thickness governs the MAWP of this nozzle.

UG-37 Area Calculation Summary (in ²) For P = 160.01 psi @ 650 °F The opening is adequately reinforced							UG-45 Nozzle Wall Thickness Summary (in) The nozzle passes UG-45	
A required	A available	A ₁	A ₂	A ₃	A ₅	A welds	t _{req}	t _{min}
4.3951	5.2553	--	--	--	5.1719	0.0834	0.2556	4.75

UG-41 Weld Failure Path Analysis Summary (lb _f) All failure paths are stronger than the applicable weld loads				
Weld load W	Weld load W ₁₋₁	Path 1-1 strength	Weld load W ₂₋₂	Path 2-2 strength
82.627.05	98.799.55	116.985.79	--	--

UW-16 Weld Sizing Summary			
Weld description	Required weld throat size (in)	Actual weld throat size (in)	Status
Pad to shell fillet (Leg ₄₂)	0.115	0.1312	weld size is adequate
Nozzle to inside shell fillet (Leg ₄₃)	0.161	0.161 (corroded)	weld size is adequate

Calculations for internal pressure 160.01 psi @ 650 °F

Parallel Limit of reinforcement per UG-40

$$\begin{aligned}
 L_R &= \text{MAX}(d, R_n + (t_n - C_n) + (t - C)) \\
 &= \text{MAX}(16.3372, 8.1686 + (0) + (0.23 - 0)) \\
 &= 16.3372 \text{ in}
 \end{aligned}$$

Outer Normal Limit of reinforcement per UG-40

$$\begin{aligned}
 L_H &= \text{MIN}(2.5*(t - C), 2.5*(t_n - C_n) + t_e) \\
 &= \text{MIN}(2.5*(0.23 - 0), 2.5*(0) + 2.44) \\
 &= 0.575 \text{ in}
 \end{aligned}$$

Required thickness t_r from UG-37(a)(c)

$$\begin{aligned}
 t_r &= P \cdot K_1 \cdot D / (2 \cdot S \cdot E - 0.2 \cdot P) \\
 &= 160.0094 \cdot 0.9 \cdot 60 / (2 \cdot 18,800 \cdot 1 - 0.2 \cdot 160.0094) \\
 &= 0.23 \text{ in}
 \end{aligned}$$

Area required per UG-37(c)

Allowable stresses: $S_n = 17,800$, $S_v = 18,800$, $S_p = 17,800$ psi

$$f_{r1} = 1$$

$$f_{r2} = \text{lesser of } 1 \text{ or } S_n / S_v = 0.9468$$

$$f_{r4} = \text{lesser of } 1 \text{ or } S_p / S_v = 0.9468$$

$$\begin{aligned} A &= d \cdot t_r \cdot F + 2 \cdot t_n \cdot t_r \cdot F \cdot (1 - f_{r1}) + \text{Tapped hole area loss} \\ &= 16.3372 \cdot 0.23 \cdot 1 + 2 \cdot 0 \cdot 0.23 \cdot 1 \cdot (1 - 1) + 0.6375 \\ &= \underline{4.3951} \text{ in}^2 \end{aligned}$$

Area available from FIG. UG-37.1

$$A_1 = \text{larger of the following} = \underline{0} \text{ in}^2$$

$$\begin{aligned} &= d \cdot (E_1 \cdot t - F \cdot t_r) - 2 \cdot t_n \cdot (E_1 \cdot t - F \cdot t_r) \cdot (1 - f_{r1}) \\ &= 16.3372 \cdot (1 \cdot 0.23 - 1 \cdot 0.23) - 2 \cdot 0 \cdot (1 \cdot 0.23 - 1 \cdot 0.23) \cdot (1 - 1) \\ &= 0 \text{ in}^2 \\ &= 2 \cdot (t + t_n) \cdot (E_1 \cdot t - F \cdot t_r) - 2 \cdot t_n \cdot (E_1 \cdot t - F \cdot t_r) \cdot (1 - f_{r1}) \\ &= 2 \cdot (0.23 + 0) \cdot (1 \cdot 0.23 - 1 \cdot 0.23) - 2 \cdot 0 \cdot (1 \cdot 0.23 - 1 \cdot 0.23) \cdot (1 - 1) \\ &= 0 \text{ in}^2 \end{aligned}$$

Area A_2 is not included in these calculations.

$$\begin{aligned} A_{42} &= \text{Leg}^2 \cdot f_{r4} \\ &= 0.1875^2 \cdot 0.9468 \\ &= \underline{0.0333} \text{ in}^2 \end{aligned}$$

$$\begin{aligned} A_{43} &= \text{Leg}^2 \cdot f_{r2} \\ &= 0.23^2 \cdot 0.9468 \\ &= \underline{0.0501} \text{ in}^2 \end{aligned}$$

$$\begin{aligned} A_5 &= (D_p - \text{Pad ID}) \cdot t_e \cdot f_{r4} \\ &= (25.5 - 16) \cdot 0.575 \cdot 0.9468 \\ &= \underline{5.1719} \text{ in}^2 \end{aligned}$$

$$\begin{aligned} \text{Area} &= A_1 + A_{42} + A_{43} + A_5 \\ &= 0 + 0.0333 + 0.0501 + 5.1719 \\ &= \underline{5.2553} \text{ in}^2 \end{aligned}$$

As $\text{Area} \geq A$ the reinforcement is adequate.

UW-16(d)(2) Weld Check

$$\begin{aligned}\text{Inner fillet: } t_{\min} &= \text{lesser of } 0.75 \text{ or } t_e \text{ or } t = 0.23 \text{ in} \\ t_{w(\min)} &= 0.5 * t_{\min} = \underline{0.115} \text{ in} \\ t_{w(\text{actual})} &= 0.7 * \text{Leg} = 0.7 * 0.1875 = 0.1313 \text{ in}\end{aligned}$$

The fillet weld size is satisfactory.

$$\begin{aligned}\text{Lower fillet: } t_{w(\min)} &= 0.7 * t_{\min} = \underline{0.161} \text{ in} \\ t_{w(\text{actual})} &= 0.7 * \text{Leg} = 0.7 * 0.23 = 0.161 \text{ in}\end{aligned}$$

The fillet weld size is satisfactory.

UG-45 Nozzle Neck Thickness Check

Interpretation VIII-1-83-66 has been applied.

$$\begin{aligned}t_{a \text{ UG-27}} &= P * R / (S * E - 0.6 * P) + \text{Corrosion} \\ &= 160.0094 * 8.1686 / (17,800 * 1 - 0.6 * 160.0094) + 0 \\ &= 0.0738 \text{ in}\end{aligned}$$

$$\begin{aligned}t_a &= \max[t_{a \text{ UG-27}}, t_{a \text{ UG-22}}] \\ &= \max[0.0738, 0] \\ &= 0.0738 \text{ in}\end{aligned}$$

$$t_{b1} = 0.2556 \text{ in}$$

$$\begin{aligned}t_{b1} &= \max[t_{b1}, t_{b \text{ UG16}}] \\ &= \max[0.2556, 0.0625] \\ &= 0.2556 \text{ in}\end{aligned}$$

$$\begin{aligned}t_b &= \min[t_{b3}, t_{b1}] \\ &= \min[0.3281, 0.2556] \\ &= 0.2556 \text{ in}\end{aligned}$$

$$\begin{aligned}t_{\text{UG-45}} &= \max[t_a, t_b] \\ &= \max[0.0738, 0.2556] \\ &= 0.2556 \text{ in}\end{aligned}$$

Available nozzle wall thickness new, $t_n = 4.75$ in

The nozzle neck thickness is adequate.

Allowable stresses in joints UG-45 and UW-15(c)

Outer fillet weld in shear: $0.49 * 17,800 = 8,722$ psi

Lower fillet weld in shear: $0.49 * 17,800 = 8,722$ psi

(1) Outer fillet weld in shear

$$(\pi / 2) * \text{Pad OD} * \text{Leg} * S_o = (\pi / 2) * 25.5 * 0.1875 * 8,722 = 65,505.45 \text{ lb}_f$$

(2) Lower fillet weld in shear

$$(\pi / 2) * 16.3372 * \text{Leg} * S_l = (\pi / 2) * 16.3372 * 0.23 * 8,722 = 51,480.34 \text{ lb}_f$$

Loading on welds per UG-41(b)(1)

$$\begin{aligned} W &= (A - A_1 + 2 * t_n * f_{r1} * (E_1 * t - F * t_r)) * S_v \\ &= (4.3951 - 0 + 2 * 0 * 1 * (1 * 0.23 - 1 * 0.23)) * 18,800 \\ &= \underline{82,627.05} \text{ lb}_f \end{aligned}$$

$$\begin{aligned} W_{1-1} &= (A_5 + A_{42} + A_{43}) * S_v \\ &= (5.1719 + 0.0333 + 0.0501) * 18,800 \\ &= \underline{98,799.55} \text{ lb}_f \end{aligned}$$

Load for path 1-1 lesser of W or $W_{1-1} = 82,627.05 \text{ lb}_f$

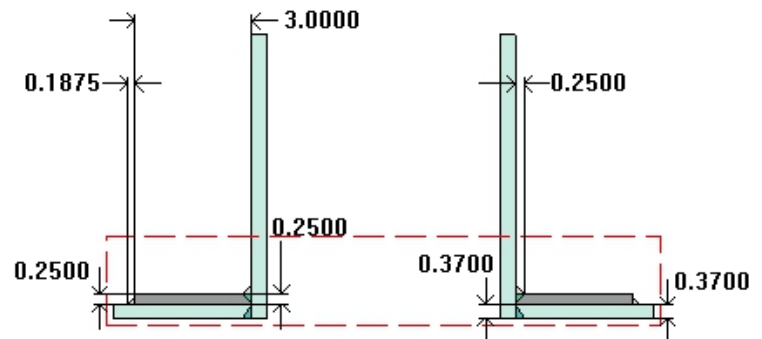
Path 1-1 through (1) & (2) = $65,505.45 + 51,480.34 = \underline{116,985.79} \text{ lb}_f$

Path 1-1 is stronger than W so it is acceptable per UG-41(b)(2).

Nozzle #3 (N3)

ASME Section VIII Division 1, 2013 Edition (Code Case 2695)					
Component	Nozzle				
Nozzle Material	SA-106 B Smls pipe (II-D p. 14, ln. 19)				
Pad Material	SA-516 70 (II-D p. 22, ln. 6)				
	Impact Tested	Normalized	Fine Grain Practice	PWHT	Optimize MDMT/ Find MAWP
Nozzle	No	No	No	No	No
Pad	No	No	No	No	No
	Design Pressure (psi)		Design Temperature (°F)		Design MDMT (°F)
Internal	100		650		-20
Location and Orientation					
Attached to	Cylinder #2				
Orientation	hillside				
Offset, L	48"				
Angle, θ	0°				
Distance, r	48"				
Offset CL to CL, L_o	15"				
Through a Category A Joint	No				

Dimensions		
Description	NPS 12 Std Weight	
Inside Diameter	12"	
Nominal Wall Thickness	0.375"	
Local Vessel Minimum Thickness	0.37"	
Nozzle to Pad Groove Weld, t_{w2}	0.25"	
Leg ₄₁	0.25"	
Leg ₄₂	0.1875"	
Pad Width, W	3"	
Pad Thickness, t_e	0.25"	
External Projection Available, L_{pr1}	17.0237"	
Projection outside vessel to flange face, L_f	18.8362"	
Corrosion	Inner	0"
	Outer	0"



Summary Tables

Division 2 4.5.6 Maximum local primary membrane stress					
Condition	P (psi)	P _L (psi)	S _{allow} (psi)	Over stressed	P _{max} (psi)
Design P	100	16.021	28.200	No	176.02

Division 2 4.5.4.1 Nozzle Neck Minimum Thickness								
Condition	P (psi)	t _{d1} (in)	t _{d,STD} (in)	t (in)	t _d (in)	t _n (in)	Status	P _{max} (psi)
Design P	100	0.0352	0.3281	0.37	0.3281	0.375	OK	1.036.68

Division 2 4.5.14 Strength of Nozzle Attachment Welds Summary											
Condition	P (psi)	k _y	L _τ (in)	L _{τp} (in)	f _{welds} (lb _f)	τ ₁ (psi)	τ ₂ (psi)	τ ₃ (psi)	τ (psi)	S (psi)	Over stressed
Design P	100	1.0625	10.0138	14.7262	3.191.91	857	543	1.345	1.345	18,800	No
MAWP	176.02	1.0625	10.0138	14.7262	5.618.38	1.508	956	2.368	2.368	18,800	No

Division 2 Table 4.2.11 Minimum Fillet Weld Sizes			
Fillet Leg	Required Leg Size (in)	Actual Leg Size (in)	Status
Leg ₄₁	0.3571	0.25	weld size is NOT adequate
Leg ₄₂	0.2143	0.1875	weld size is NOT adequate

Detailed Calculations

Division 2 4.5.6 Vessel effective radius	
$R_{\text{eff}} = 0.5 \cdot D_i$	(4.5.3)
$R_{\text{xn}} = t_n / \ln[1 + t_n / R_n]$	(4.5.51)
Corroded\New	
$R_{\text{eff}} = 0.5 \cdot 60.085 =$	30.0425"
$R_{\text{xn}} = 0.375 / \ln[1 + 0.375 / 6] =$	6.1856"

Division 2 4.5.6 Radius of nozzle opening along the long chord	
$D_x = \text{Abs}(L_o)$	
$\theta_1 = \arccos[D_x / R_{\text{eff}}]$	(4.5.65)
$\theta_2 = \arccos[(D_x + R_n) / R_{\text{eff}}]$	(4.5.66)
$R_{\text{ncl}} = R_{\text{eff}} \cdot (\theta_1 - \theta_2) \cdot \pi / 180$	(4.5.64)
$R_{\text{nc}} = \max[(R_{\text{ncl}} / 2), R_n]$	(4.5.63)
Corroded\New	
$D_x = \text{Abs}(L_o) =$	15"
$\theta_1 = \arccos[15 / 30.0425] =$	60.0468°
$\theta_2 = \arccos[(15 + 6) / 30.0425] =$	45.6524°
$R_{\text{ncl}} = 30.0425 \cdot (60.0468 - 45.6524) \cdot \pi / 180 =$	7.5476"
$R_{\text{nc}} = \max[(7.5476 / 2), 6] =$	6"

Division 2 4.5.6 Limit of reinforcement along the vessel wall	
$L_{R1} = (R_{\text{eff}} \cdot t)^{0.5} + W$	(4.5.6)
$L_{R2} = ((R_{\text{eff}} + t) \cdot (t + t_e))^{0.5}$	(4.5.7)
$L_{R3} = 2 \cdot R_n$	(4.5.8)
$L_R = \min[L_{R1}, L_{R2}, L_{R3}]$	(4.5.9)
Corroded\New	
$L_{R1} = (30.0425 \cdot 0.37)^{0.5} + 3 =$	6.334"
$L_{R2} = ((30.0425 + 0.37) \cdot (0.37 + 0.25))^{0.5} =$	4.3423"
$L_{R3} = 2 \cdot 6 =$	12"
$L_R = \min[6.334, 4.3423, 12] =$	4.3423"

Division 2 4.5.6 Limit of reinforcement along the nozzle wall projecting outside the vessel surface	
$L_{H1} = \min[1.5 \cdot t, t_e] + (R_n \cdot t_n)^{0.5}$	(4.5.11)
$L_{H2} = L_{pr1}$	(4.5.12)
$L_{H3} = 8 \cdot (t + t_e)$	(4.5.13)
$L_H = \min[L_{H1}, L_{H2}, L_{H3}] + t$	(4.5.14)
Corroded\New	
$L_{H1} = \min[1.5 \cdot 0.37, 0.25] + (6 \cdot 0.375)^{0.5} =$	1.75"
$L_{H2} = 17.0237 =$	17.0237"
$L_{H3} = 8 \cdot (0.37 + 0.25) =$	4.96"
$L_H = \min[1.75, 17.0237, 4.96] + 0.37 =$	2.12"

Division 2 4.5.6 Available area near the nozzle opening	
$\lambda = \min[(2 \cdot R_n + t_n) / ((D_i + t_{eff}) \cdot t_{eff})^{0.5}, 12]$	(4.5.24)
$A_1 = t \cdot L_R \cdot \max[(\lambda / 5)^{0.85}, 1]$	(4.5.23)
$A_2 = t_n \cdot L_H$	(4.5.27)
$A_{41} = 0.5 \cdot L_{41}^2$	(4.5.39)
$A_{42} = 0.5 \cdot L_{42}^2$	(4.5.40)
$A_{5a} = W \cdot t_e$	(4.5.42)
$A_{5b} = L_R \cdot t_e$	(4.5.43)
$A_5 = \min[A_{5a}, A_{5b}]$	(4.5.45)
Corroded\New	
$\lambda = \min[(2 \cdot 6 + 0.375) / ((60.085 + 0.5427) \cdot 0.5427)^{0.5}, 12] =$	2.1574
$A_1 = 0.37 \cdot 4.3423 \cdot \max[(2.1574 / 5)^{0.85}, 1] =$	1.6067 in ²
$A_2 = 0.375 \cdot 2.12 =$	0.795 in ²
$A_{41} = 0.5 \cdot 0.25^2 =$	0.0313 in ²
$A_{42} = 0.5 \cdot 0.1875^2 =$	0.0176 in ²
$A_{5a} = 3 \cdot 0.25 =$	0.75 in ²
$A_{5b} = 4.3423 \cdot 0.25 =$	1.0856 in ²
$A_5 = \min[0.75, 1.0856] =$	0.75 in ²

Division 2 4.5.6 Equations	
$f_m = \min[S_n / S , 1]$	(4.5.21)
$f_{rp} = \min[S_p / S , 1]$	(4.5.22)
$t_{eff} = t + (A_5 * f_{rp} / L_R)$	(4.5.25)
$R_{xs} = t_{eff} / \ln[1 + t_{eff} / R_{eff}]$	(4.5.52)
$A_T = A_1 + f_m * A_2 + A_{41} + A_{42} + f_{rp} * A_5$	(4.5.20)
$f_N = P * R_{xn} * L_H$	(4.5.46)
$f_S = P * R_{xs} * (L_R + t_n)$	(4.5.48)
$f_Y = P * R_{xs} * R_{nc}$	(4.5.50)
$\sigma_{avg} = (f_N + f_S + f_Y) / A_T$	(4.5.53)
$\sigma_{circ} = P * R_{xs} / t_{eff}$	(4.5.54)
$P_L = \max[\{2 * \sigma_{avg} - \sigma_{circ}\} , \sigma_{circ}]$	(4.5.55)
$S_{allow} = 1.5 * S * E$	(4.5.57)
$P_L \leq S_{allow}$	(4.5.56)
$A_p = (f_N + f_S + f_Y) / P$	(4.5.62)
$P_{max1} = S_{allow} / (2 * A_p / A_T - R_{xs} / t_{eff})$	(4.5.59)
$P_{max2} = S * (t / R_{xs})$	(4.5.60)
$P_{max} = \min[P_{max1} , P_{max2}]$	(4.5.61)

Division 2 4.5.6 Design Rules for Openings in Shells and Heads	
Design P	
Total available area near the nozzle opening	
$f_m = \min[17,100 / 18,800 , 1] =$	0.9096
$f_{rp} = \min[18,800 / 18,800 , 1] =$	1
$t_{eff} = 0.37 + (0.75 \cdot 1 / 4.3423) =$	0.5427"
$R_{xs} = 0.5427 / \ln[1 + 0.5427 / 30.0425] =$	30.3131"
$A_T = 1.6067 + 0.9096 \cdot 0.795 + 0.0313 + 0.0176 + 1 \cdot 0.75 =$	3.1286 in ²
Forces at nozzle to vessel intersection	
$f_N = 100 \cdot 6.1856 \cdot 2.12 =$	1,311.35 lb _f
$f_S = 100 \cdot 30.3131 \cdot (4.3423 + 0.375) =$	14,299.64 lb _f
$f_Y = 100 \cdot 30.3131 \cdot 6 =$	18,187.83 lb _f
Average local primary membrane stress	
$\sigma_{avg} = (1,311.35 + 14,299.64 + 18,187.83) / 3.1286 =$	10,803 psi
General primary membrane stress	
$\sigma_{circ} = 100 \cdot 30.3131 / 0.5427 =$	5,585 psi
Maximum local primary membrane stress at the nozzle intersection	
$P_L = \max[\{2 \cdot 10,803 - 5,585\} , 5,585] =$	16,021 psi
Allowable stress	
$S_{allow} = 1.5 \cdot 18,800 \cdot 1 =$	28,200 psi
$P_L = 16,021 \text{ psi} \leq S_{allow} = 28,200 \text{ psi}$	
Maximum allowable working pressure	
$A_p = (1,311.35 + 14,299.64 + 18,187.83) / 100 =$	337.9882 in ²
$P_{max1} = 28,200 / (2 \cdot 337.9882 / 3.1286 - 30.3131 / 0.5427) =$	176.02 psi
$P_{max2} = 18,800 \cdot (0.37 / 30.3131) =$	229.47 psi
$P_{max} = \min[176.02, 229.47] =$	176.02 psi

Division 2 4.5.14 Equations	
$k_y = (R_{nc} + t_n) / R_{nc}$	(4.5.174)
$L_\tau = \pi / 2 * [((R_{nc} + t_n)^2 + (R_n + t_n)^2) / 2]^{0.5}$	(4.5.176)
$L_{tp} = \pi / 2 * [((R_{nc} + t_n + W)^2 + (R_n + t_n + W)^2) / 2]^{0.5}$	(4.5.178)
$L_{41T} = 0.7071 * L_{41}$	(4.5.179)
$L_{42T} = 0.7071 * L_{42}$	(4.5.180)
$f_{welds} = \min[f_y * k_y, 1.5 * S_n * (A_2 + A_3), \pi / 4 * P * R_n^2 * k_y^2]$	(4.5.184)
$f_{ws} = f_{welds} * t * S / (t * S + t_e * S_p)$	(4.5.189)
$f_{wp} = f_{welds} * t_e * S_p / (t * S + t_e * S_p)$	(4.5.190)
$\tau_1 = f_{ws} / [L_\tau * (0.6 * t_{w1} + 0.49 * L_{43T})]$	(4.5.185)
$\tau_2 = f_{wp} / [L_\tau * (0.6 * t_{w2} + 0.49 * L_{41T})]$	(4.5.186)
$\tau_3 = f_{wp} / [L_{tp} * 0.49 * L_{42T}]$	(4.5.187)
$\tau = \max[\tau_1, \tau_2, \tau_3]$	
$\max[\tau_1, \tau_2, \tau_3] \leq S$	(4.5.188)

Division 2 4.5.14 Strength of Nozzle Attachment Welds	
Corroded\New	
Discontinuity force factor	
$k_y = (6 + 0.375) / 6 =$	1.0625
Weld length resisting discontinuity force	
$L_\tau = \pi / 2 * [(6 + 0.375)^2 + (6 + 0.375)^2] / 2^{0.5} =$	10.0138"
$L_{\tau p} = \pi / 2 * [(6 + 0.375 + 3)^2 + (6 + 0.375 + 3)^2] / 2^{0.5} =$	14.7262"
Weld throat dimensions	
$L_{41T} = 0.7071 * 0.25 =$	0.1768"
$L_{42T} = 0.7071 * 0.1875 =$	0.1326"
Design P	
Average shear stress in weld	
$f_{welds} = \min[18,187.83 * 1.0625, 1.5 * 17,100 * (0.795 + 0) , \pi / 4 * 100 * 6^2 * 1.0625^2] =$	3,191.91 lb_f
$f_{ws} = 3,191.91 * 0.37 * 18,800 / (0.37 * 18,800 + 0.25 * 18,800) =$	1,904.85 lb _f
$f_{wp} = 3,191.91 * 0.25 * 18,800 / (0.37 * 18,800 + 0.25 * 18,800) =$	1,287.06 lb _f
$\tau_1 = 1,904.85 / [10.0138 * (0.6 * 0.37 + 0.49 * 0)] =$	857 psi
$\tau_2 = 1,287.06 / [10.0138 * (0.6 * 0.25 + 0.49 * 0.1768)] =$	543 psi
$\tau_3 = 1,287.06 / [14.7262 * 0.49 * 0.1326] =$	1,345 psi
$\tau = \max[857, 543, 1,345] =$	1,345 psi
$\tau = 1,345 \text{ psi} \leq S = 18,800 \text{ psi}$	
MAWP	
Average shear stress in weld	
$f_{welds} = \min[32,014.12 * 1.0625, 1.5 * 17,100 * (0.795 + 0) , \pi / 4 * 176.02 * 6^2 * 1.0625^2] =$	5,618.38 lb_f
$f_{ws} = 5,618.38 * 0.37 * 18,800 / (0.37 * 18,800 + 0.25 * 18,800) =$	3,352.9 lb _f
$f_{wp} = 5,618.38 * 0.25 * 18,800 / (0.37 * 18,800 + 0.25 * 18,800) =$	2,265.48 lb _f
$\tau_1 = 3,352.9 / [10.0138 * (0.6 * 0.37 + 0.49 * 0)] =$	1,508 psi
$\tau_2 = 2,265.48 / [10.0138 * (0.6 * 0.25 + 0.49 * 0.1768)] =$	956 psi
$\tau_3 = 2,265.48 / [14.7262 * 0.49 * 0.1326] =$	2,368 psi
$\tau = \max[1,508, 956, 2,368] =$	2,368 psi
$\tau = 2,368 \text{ psi} \leq S = 18,800 \text{ psi}$	

Division 2 4.3.3.1 Design Thickness for Internal Pressure	
$t_d = D / 2 * (\exp[P / (S * E)] - 1) + \text{Corrosion}$	(4.3.1)
Operating Hot & Corroded	
$t_d = 12 / 2 * (\exp[(100 + 0) / (17,100 * 1)] - 1) + 0 =$	0.0352"
Operating Cold & Corroded	
$t_r = 12 / 2 * (\exp[(144.02 + 0) / (17,100 * 1)] - 1) =$	0.0507"

Division 2 4.3.3.1 Maximum Pressure Circumferential Stress	
$P_{\max} = S * E * \ln[D_o / D]$	(4.3.1)
MAWP	
$P_{\max} = 17,100 * 1 * \ln[12.75 / 12] - 0 =$	1,036.68 psi

Saddle #1

ASME Section VIII Division 1, 2013 Edition (Code Case 2695)		
Saddle Material		
Saddle Construction	Centered web	
Welded to Vessel	Yes	
Saddle Allowable Stress	20,000 psi	
Saddle Yield Stress	38,000 psi	
Foundation Allowable Stress	1,658 psi	
Design Pressure	Left Saddle	Right Saddle
Operating	144.02 psi	
Test	4.12 psi	7.59 psi
Dimensions		
Saddle distance to datum	13.1875"	
Tangent To Tangent Length, L	196"	
Saddle seperation, L _s	165.625"	
Vessel Mean Radius, R _m	30.3145"	
Tangent Distance Left, a _l	15.1875"	
Tangent Distance Right, a _r	15.1875"	
Saddle Height, H _s	42.375"	
Saddle Contact Angle, θ	120°	
Web Plate Thickness, t _s	0.375"	
Base Plate Length, E	54.0667"	
Base Plate Width, F	9"	
Base Plate Thickness, t _b	0.375"	
Number of Stiffening Ribs, n	4	
Largest Stiffening Rib Spacing, d _i	17.7514"	
Stiffening Rib Thickness, t _w	0.375"	
Saddle Width, b	8"	
Reinforcing Plate		
Material	SA-516 70 (II-D p. 22, In. 6)	
Thickness, t _r	0.375"	
Width, b ₁	11"	
Contact Angle, θ ₁	132°	
Bolting		

Material		
Bolt Allowable Shear	15,000 psi	
Description	1 inch series 8 threaded	
Corrosion on root	0"	
Anchor Bolts per Saddle	2	
Base coefficient of friction	0.45	
Weight		
	Operating, Corroded	Hydrotest
Weight on Left Saddle	2,521 lb	13,700 lb
Weight on Right Saddle	2,310 lb	13,456 lb
Weight of Saddle Pair	578 lb	

Reinforcing Plate Requirements		
$b_1 \geq \min[(b + 1.56 \cdot (R_m \cdot t)^{0.5}), 2 \cdot a]$	(4.15.1)	
$\theta_1 \geq \theta + \theta / 12$	(4.15.2)	
Operating		
$b_1 \geq \min[(8 + 1.56 \cdot (30.3145 \cdot 0.371)^{0.5}), 2 \cdot 15.1875]$	13.2316"	Not OK
$\theta_1 \geq 120 + 120 / 12$	130 °	OK

4.15.3.3 Longitudinal Stress										
Load	Condition	Saddle	Bending + Pressure at Saddle (psi)				Bending + Pressure between Saddle (psi)			
			σ_3^*	Allowable	σ_4^*	Allowable	σ_1	Allowable	σ_2	Allowable
Weight	Operating	Right Saddle	5.903	18,800	5.872	18,800	5.962	18,800	5.805	18,800
		Left Saddle	5.903	18,800	5.873	18,800				
	Test	Right Saddle	420	34,200	249	34,200	713	34,200	-93	30,364
		Left Saddle	276	34,200	109	34,200				
	Hot Shutdown	Right Saddle	20	18,800	-11	18,800	79	18,800	-79	18,800
		Left Saddle	19	18,800	-11	18,800				

4.15.3.4 Shear Stress				
Load	Condition	Saddle	Shear Stress in Cylindrical Shell (psi)	
			τ_2	Allowable
Weight	Operating	Right Saddle	183	15,040
		Left Saddle	204	15,040
	Test	Right Saddle	1,075	27,360
		Left Saddle	1,095	27,360

4.15.3.5 Circumferential Stress						
Load	Condition	Saddle	Compressive Circ Membrane Stress In Shell (psi)		Compressive Membrane + Bending Stress at Horn (psi)	
			σ_6	Allowable	σ_7^*	Allowable
Weight	Operating	Right Saddle	-36	18,800	-537	23,500
		Left Saddle	-39	18,800	-580	23,500
	Test	Right Saddle	-209	34,200	-3,127	34,200
		Left Saddle	-212	34,200	-3,153	34,200

4.15.3.3 Longitudinal Stress				
$M_1 = w^*(2*h_2*a / 3 + a^2 / 2 - (R_m^2 - h_2^2) / 4)$				
$M_2 = w^*[a^2 / 2 + 2 / 3*h_2*(3 / 8*h_2 + a) - x_0^2 / 2 - R_m^2 / 4]$				
$\sigma_1 = P*R_m / (2*t) - M_2 / (\pi*R_m^2*t)$				(4.15.6)
$\sigma_2 = P*R_m / (2*t) + M_2 / (\pi*R_m^2*t)$				(4.15.7)
$\sigma_3^* = P*R_m / (2*t) - M_1 / (\pi*K_1*R_m^2*t)$				(4.15.10)
$\sigma_4^* = P*R_m / (2*t) + M_1 / (\pi*K_1*R_m^2*t)$				(4.15.11)
$S_c = \min[K*t*E_y / (16*R_m) , S^*E]$				(4.15.12)
Weight				
Operating	Right Saddle	$M_1 = 22.33*(2*15.23*15.1875 / 3 + 15.1875^2 / 2 - (30.2275^2 - 15.23^2) / 4)$	-2,213.2 lb _f -in	
		$S_c = \min[1*0.37*26.0E+06 / (16*30.2275) , 18,800*1]$	18,800 psi	
			Stress (psi)	Allowable (psi)
		$\sigma_3^* = 144.02*30.2275 / (2*0.37) - -2,213.2 / (\pi*0.1066*30.2275^2*0.37)$	5.903	18,800
		$\sigma_4^* = 144.02*30.2275 / (2*0.37) + -2,213.2 / (\pi*0.1923*30.2275^2*0.37)$	5.872	18,800
	Between Saddles	$x_0 = 2,521 / 22.33 - 2*15.23 / 3 - 15.1875$	87.5362"	
		$M_2 = 22.33*[15.1875^2 / 2 + 2 / 3*15.23*(3 / 8*15.23 + 15.1875) - 87.5362^2 / 2 - 30.2275^2 / 4]$	-83,384.5 lb _f -in	
		$S_c = \min[1*0.37*26.0E+06 / (16*30.2275) , 18,800*1]$	18,800 psi	
			Stress (psi)	Allowable (psi)
		$\sigma_1 = 144.02*30.2275 / (2*0.37) - -83,384.5 / (\pi*30.2275^2*0.37)$	5.962	18,800
		$\sigma_2 = 144.02*30.2275 / (2*0.37) + -83,384.5 / (\pi*30.2275^2*0.37)$	5.805	18,800
	Left Saddle	$M_1 = 22.33*(2*15.23*15.1875 / 3 + 15.1875^2 / 2 - (30.3145^2 - 15.23^2) / 4)$	-2,183.8 lb _f -in	
		$S_c = \min[1*0.371*26.0E+06 / (16*30.3145) , 18,800*1]$	18,800 psi	
			Stress (psi)	Allowable (psi)
		$\sigma_3^* = 144.02*30.3145 / (2*0.371) - -2,183.8 / (\pi*0.1066*30.3145^2*0.371)$	5.903	18,800
		$\sigma_4^* = 144.02*30.3145 / (2*0.371) + -2,183.8 / (\pi*0.1923*30.3145^2*0.371)$	5.873	18,800
	Right Saddle	$M_1 = 125.54*(2*15.23*15.1875 / 3 + 15.1875^2 / 2 - (30.2275^2 - 15.23^2) / 4)$	-12,441 lb _f -in	
		$S_c = \min[1.35*0.37*29.4E+06 / (16*30.2275) , 34,200*1]$	30,364 psi	

			Stress (psi)	Allowable (psi)
		$\sigma_3^* = 7.59 \cdot 30.2275 / (2 \cdot 0.37) - -12,441 / (\pi \cdot 0.1066 \cdot 30.2275^2 \cdot 0.37)$	420	34,200
		$\sigma_4^* = 7.59 \cdot 30.2275 / (2 \cdot 0.37) + -12,441 / (\pi \cdot 0.1923 \cdot 30.2275^2 \cdot 0.37)$	249	34,200
		$x_0 = 13,700 / 125.54 - 2 \cdot 15.23 / 3 - 15.1875$	83.7843"	
		$M_2 = 125.54 \cdot [15.1875^2 / 2 + 2 / 3 \cdot 15.23 \cdot (3 / 8 \cdot 15.23 + 15.1875) - 83.7843^2 / 2 - 30.2275^2 / 4]$	-428,371.4 lb _f -in	
		$S_c = \min[1.35 \cdot 0.37 \cdot 29.4E+06 / (16 \cdot 30.2275) , 34,200 \cdot 1]$	30,364 psi	
	Between Saddles		Stress (psi)	Allowable (psi)
		$\sigma_1 = 7.59 \cdot 30.2275 / (2 \cdot 0.37) - -428,371.4 / (\pi \cdot 30.2275^2 \cdot 0.37)$	713	34,200
		$\sigma_2 = 7.59 \cdot 30.2275 / (2 \cdot 0.37) + -428,371.4 / (\pi \cdot 30.2275^2 \cdot 0.37)$	-93	30,364
		$M_1 = 125.54 \cdot (2 \cdot 15.23 \cdot 15.1875 / 3 + 15.1875^2 / 2 - (30.3145^2 - 15.23^2) / 4)$	-12,275.7 lb _f -in	
		$S_c = \min[1.35 \cdot 0.371 \cdot 29.4E+06 / (16 \cdot 30.3145) , 34,200 \cdot 1]$	30,359 psi	
			Stress (psi)	Allowable (psi)
	Left Saddle	$\sigma_3^* = 4.12 \cdot 30.3145 / (2 \cdot 0.371) - -12,275.7 / (\pi \cdot 0.1066 \cdot 30.3145^2 \cdot 0.371)$	276	34,200
		$\sigma_4^* = 4.12 \cdot 30.3145 / (2 \cdot 0.371) + -12,275.7 / (\pi \cdot 0.1923 \cdot 30.3145^2 \cdot 0.371)$	109	34,200
	Right Saddle	$M_1 = 22.33 \cdot (2 \cdot 15.23 \cdot 15.1875 / 3 + 15.1875^2 / 2 - (30.2275^2 - 15.23^2) / 4)$	-2,213.2 lb _f -in	
		$S_c = \min[1 \cdot 0.37 \cdot 26.0E+06 / (16 \cdot 30.2275) , 18,800 \cdot 1]$	18,800 psi	
			Stress (psi)	Allowable (psi)
		$\sigma_3^* = 0 \cdot 30.2275 / (2 \cdot 0.37) - -2,213.2 / (\pi \cdot 0.1066 \cdot 30.2275^2 \cdot 0.37)$	20	18,800
		$\sigma_4^* = 0 \cdot 30.2275 / (2 \cdot 0.37) + -2,213.2 / (\pi \cdot 0.1923 \cdot 30.2275^2 \cdot 0.37)$	-11	18,800
Hot Shutdown	Between Saddles	$x_0 = 2,521 / 22.33 - 2 \cdot 15.23 / 3 - 15.1875$	87.5362"	
		$M_2 = 22.33 \cdot [15.1875^2 / 2 + 2 / 3 \cdot 15.23 \cdot (3 / 8 \cdot 15.23 + 15.1875) - 87.5362^2 / 2 - 30.2275^2 / 4]$	-83,384.5 lb _f -in	
		$S_c = \min[1 \cdot 0.37 \cdot 26.0E+06 / (16 \cdot 30.2275) , 18,800 \cdot 1]$	18,800 psi	
			Stress (psi)	Allowable (psi)
		$\sigma_1 = 0 \cdot 30.2275 / (2 \cdot 0.37) - -83,384.5 / (\pi \cdot 30.2275^2 \cdot 0.37)$	79	18,800
		$\sigma_2 = 0 \cdot 30.2275 / (2 \cdot 0.37) + -83,384.5 / (\pi \cdot 30.2275^2 \cdot 0.37)$	-79	18,800
	Left Saddle	$M_1 = 22.33 \cdot (2 \cdot 15.23 \cdot 15.1875 / 3 + 15.1875^2 / 2 - (30.3145^2 - 15.23^2) / 4)$	-2,183.8 lb _f -in	

		$S_c = \min[1*0.371*26.0E+06 / (16*30.3145) , 18,800*1]$	18,800 psi	
			Stress (psi)	Allowable (psi)
		$\sigma_3^* = 0*30.3145 / (2*0.371) - 2,183.8 / (\pi*0.1066*30.3145^2*0.371)$	19	18,800
		$\sigma_4^* = 0*30.3145 / (2*0.371) + 2,183.8 / (\pi*0.1923*30.3145^2*0.371)$	-11	18,800

4.15.3.4 Shear Stress				
T = Q - w*(a + 2*h ₂ / 3)				(4.15.5)
τ ₂ = K ₂ *T / (R _m *t)				(4.15.14)
Weight				
Operating	Right Saddle	T = 2,310 - 22.33*(15.1875 + 2*15.23 / 3)	1,744.04 lb _f	
			Stress (psi)	Allowable (psi)
		τ ₂ = 1.1707*1,744.04 / (30.2275*0.37)	183	15,040
	Left Saddle	T = 2,521 - 22.33*(15.1875 + 2*15.23 / 3)	1,955.04 lb _f	
			Stress (psi)	Allowable (psi)
		τ ₂ = 1.1707*1,955.04 / (30.3145*0.371)	204	15,040
Test	Right Saddle	T = 13,456 - 125.54*(15.1875 + 2*15.23 / 3)	10,274.61 lb _f	
			Stress (psi)	Allowable (psi)
		τ ₂ = 1.1707*10,274.61 / (30.2275*0.37)	1.075	27,360
	Left Saddle	T = 13,700 - 125.54*(15.1875 + 2*15.23 / 3)	10,518.61 lb _f	
			Stress (psi)	Allowable (psi)
		τ ₂ = 1.1707*10,518.61 / (30.3145*0.371)	1.095	27,360

4.15.3.5 Circumferential Stress				
$x_1, x_2 = 0.78 \cdot (R_m \cdot t)^{0.5}$				(4.15.22)
$\sigma_6 = -K_5 \cdot Q \cdot k / (t \cdot (b + x_1 + x_2))$				(4.15.23)
$\sigma_7^* = -Q / (4 \cdot t \cdot (b + x_1 + x_2)) - 12 \cdot Q \cdot K_7 \cdot R_m / (L \cdot t^2)$				(4.15.25)
Weight				
Operating	Right Saddle	$x_1, x_2 = 0.78 \cdot (30.2275 \cdot 0.37)^{0.5}$	2.6085"	
			Stress (psi)	Allowable (psi)
		$\sigma_6 = -0.7603 \cdot 2,310 \cdot 0.1 / (0.37 \cdot (8 + 2.6085 + 2.6085))$	-36	18,800
		$\sigma_7^* = -2,310 / (4 \cdot 0.37 \cdot (8 + 2.6085 + 2.6085)) - 12 \cdot 2,310 \cdot 0.0134 \cdot 30.2275 / (196 \cdot 0.37^2)$	-537	23,500
	Left Saddle	$x_1, x_2 = 0.78 \cdot (30.3145 \cdot 0.371)^{0.5}$	2.6158"	
			Stress (psi)	Allowable (psi)
		$\sigma_6 = -0.7603 \cdot 2,521 \cdot 0.1 / (0.371 \cdot (8 + 2.6158 + 2.6158))$	-39	18,800
		$\sigma_7^* = -2,521 / (4 \cdot 0.371 \cdot (8 + 2.6158 + 2.6158)) - 12 \cdot 2,521 \cdot 0.0133 \cdot 30.3145 / (196 \cdot 0.371^2)$	-580	23,500
Test	Right Saddle	$x_1, x_2 = 0.78 \cdot (30.2275 \cdot 0.37)^{0.5}$	2.6085"	
			Stress (psi)	Allowable (psi)
		$\sigma_6 = -0.7603 \cdot 13,456 \cdot 0.1 / (0.37 \cdot (8 + 2.6085 + 2.6085))$	-209	34,200
		$\sigma_7^* = -13,456 / (4 \cdot 0.37 \cdot (8 + 2.6085 + 2.6085)) - 12 \cdot 13,456 \cdot 0.0134 \cdot 30.2275 / (196 \cdot 0.37^2)$	-3,127	34,200
	Left Saddle	$x_1, x_2 = 0.78 \cdot (30.3145 \cdot 0.371)^{0.5}$	2.6158"	
			Stress (psi)	Allowable (psi)
		$\sigma_6 = -0.7603 \cdot 13,700 \cdot 0.1 / (0.371 \cdot (8 + 2.6158 + 2.6158))$	-212	34,200
		$\sigma_7^* = -13,700 / (4 \cdot 0.371 \cdot (8 + 2.6158 + 2.6158)) - 12 \cdot 13,700 \cdot 0.0133 \cdot 30.3145 / (196 \cdot 0.371^2)$	-3,153	34,200

4.15.3.6 Saddle Support				
$F_h = Q*(1 + \cos[\beta] - 0.5*\sin^2[\beta]) / (\pi - \beta + \sin[\beta]*\cos[\beta])$				(4.15.42)
$A_e = (H_{eff}*t_s + t_r*b_1)$				
$\sigma_{SS} = F_h / A_e$				
Weight				
Operating	Right Saddle	$F_h = 2,310*(1 + \cos[120] - 0.5*\sin^2[120]) / (\pi - 2.0944 + \sin[120]*\cos[120])$	470.14 lb _f	
		$A_e = (10.0758*0.375 + 0.375*11)$	7.9 in ²	
			Stress (psi)	Allowable (psi)
		$\sigma_{SS} = 470.14 / 7.9$	59	13,333
	Left Saddle	$F_h = 2,521*(1 + \cos[120] - 0.5*\sin^2[120]) / (\pi - 2.0944 + \sin[120]*\cos[120])$	513.08 lb _f	
		$A_e = (10.1048*0.375 + 0.375*11)$	7.91 in ²	
			Stress (psi)	Allowable (psi)
		$\sigma_{SS} = 513.08 / 7.91$	65	13,333
Test	Right Saddle	$F_h = 13,456*(1 + \cos[120] - 0.5*\sin^2[120]) / (\pi - 2.0944 + \sin[120]*\cos[120])$	2,738.59 lb _f	
		$A_e = (10.0758*0.375 + 0.375*11)$	7.9 in ²	
			Stress (psi)	Allowable (psi)
		$\sigma_{SS} = 2,738.59 / 7.9$	347	34,200
	Left Saddle	$F_h = 13,700*(1 + \cos[120] - 0.5*\sin^2[120]) / (\pi - 2.0944 + \sin[120]*\cos[120])$	2,788.25 lb _f	
		$A_e = (10.1048*0.375 + 0.375*11)$	7.91 in ²	
			Stress (psi)	Allowable (psi)
		$\sigma_{SS} = 2,788.25 / 7.91$	352	34,200

Shear stress in anchor bolting, one end slotted	
Maximum seismic or wind base shear	0 lb _f
Thermal expansion base shear = $W*\mu = 2,810*0.45$	1,264.5 lb _f
Corroded root area for a 1 inch series 8 threaded bolt (2 per saddle)	0.551 in ²
Bolt shear stress = $1,264.5 / (0.551*1*2)$	1,147 psi
Anchor bolt stress is acceptable ($\leq 15,000$ psi)	

Shear stress in anchor bolting, transverse	
Maximum seismic or wind base shear	0 lb _f
Corroded root area for a 1 inch series 8 threaded bolt (2 per saddle)	0.551 in ²
Bolt shear stress = 0 / (0.551*2*2)	0 psi
Anchor bolt stress is acceptable (<= 15,000 psi)	

Web plate buckling check (Escoe pg 251)	
$S_c = K_i \pi^2 E / (12(1 - 0.3^2)(d_i / t_s)^2)$	
$S_c = \min[S_c , S]$	
$b_e = d_i t_s / (d_i t_s + 2 t_w (b - 1))$	
$F_b = n(A_s + 2 b_e t_s) S_c$	
Results	
$S_c = 1.28 \pi^2 29.0E+06 / (12(1 - 0.3^2)(17.7514 / 0.375)^2)$	14,972 psi
$S_c = \min[14,972 , 20,000]$	14,972 psi
$b_e = 17.7514 \cdot 0.375 / (17.7514 \cdot 0.375 + 2 \cdot 0.375 (8 - 1))$	0.5591"
$F_b = 4(2.8594 + 2 \cdot 0.5591 \cdot 0.375) \cdot 14,972$	196,355.04 lb _f
Saddle loading of 13,989 lb _f is <= F _b ; satisfactory.	

Primary bending + axial stress in the saddle due to end loads (assumes one saddle slotted)	
$\sigma_b = V(H_s - x_o) \cdot y / I + Q / A$	
$\sigma_b = 0(42.375 - 25.1509) \cdot 4 / 64.0659 + 2,521 / 31.191$	81 psi
The primary bending + axial stress in the saddle <= 20,000 psi; satisfactory.	

Secondary bending + axial stress in the saddle due to end loads (includes thermal expansion, assumes one saddle slotted)	
$\sigma_b = V(H_s - x_o) \cdot y / I + Q / A$	
$\sigma_b = 1,264.5(42.375 - 25.1509) \cdot 4 / 64.0659 + 2,521 / 31.191$	1,441 psi
The secondary bending + axial stress in the saddle < 2*S _y = 76,000 psi; satisfactory.	

Saddle base plate thickness check (Roark sixth edition, Table 26, case 7a)	
$t_b = (\beta_1 q b^2 / (1.5 S_a))^{0.5}$	
$t_b = (3 \cdot 29 \cdot 4.3125^2 / (1.5 \cdot 20,000))^{0.5}$	0.2312"
The base plate thickness of 0.375 in is adequate.	

Foundation bearing check	
$S_f = Q_{max} / (F \cdot E)$	
$S_f = 13,989 / (9 \cdot 54.0667)$	29 psi

Concrete bearing stress $\leq 1,658$ psi ; satisfactory.